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ENGINE TESTS USING HIGH-SULFUR DIESEL FUEL

FINAL REPORT AFLRL No. 129

by

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U.S. Army Fuels and Lubricants Research Laboratory
Southwest Research Institute
San Antonio, Texas

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U.S. Army Mobility Equipment
Research and Development Command
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Fort Belvoir, Virginia

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This report covers the engine test evaluation of for its effectiveness in combating the deleteric sulfur diesel fuel in a two-cycle U.S. Army dies also covers the 6V-53T testing of a preservative previous testing had shown promise in controlling high-sulfur fuel.	an organo-zinc additive ous effects of using high- sel engine. The report engine oil which in
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The fuel additive helped control fire-ring wear when high-sulfur fuel was used; however, it is not recommended for use because it caused a power loss due to fuel injector loading and the increased fuel ash (due to the additive) was contributing to exhaust valve distress.

When tested in the 6V-53T engine, the preservative engine oil/ high-sulfur fuel combination had more fire-ring wear than the reference MIL-L-2104C oil/low-sulfur fuel combination. In other performance areas, these two tests were very similar. Results of a test using reference MIL-L-2104C oil and high-sulfur fuel were inconclusive due to operational-mechanical problems encountered during the test.

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FOREWORD

The work reported herein was conducted at the U.S. Army Fuels and Lubricants Research Laboratory (USAFLRL), located at Southwest Research Institute, San Antonio, Texas under Contract DAAK70-79-C-0215. The Contracting Officer's representative was Dr. James V. Mengenhauser, DRMDE-GL, of USAMERADCOM. Mr. T. C. Bowen of the same office was project technical monitor.

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I. INTRODUCTION

A single family of high output two-stroke cycle diesel engines is used in a significant portion of the U.S. Army Combat/Tactical Fleet. Table 1 provides a listing of vehicles utilizing this engine family. The engine manufacturer

TABLE 1. ARMY TACTICAL VEHICLES POWERED BY TWO-CYCLE DIESEL ENGINES

Designation	Description	Engine Model
M106A1	Mortar, Self-propelled. 107mm	6V-53
M107	Gun, Self-propelled. 175mm	8V-71T
M108	Howitzer. Self-propelled. 105mm	8V-71T
M109	Howitzer. Medium. 155mm	8V-71T
M110	Howitzer. Self-propelled.	8V-71T
M113A1	Carrier. Personnel	6V~53
M125A1	Mortar. Self-propelled. Full-tracked	6V ~ 53
M132A1	Flame Thrower. Self-propelled.	6V-53
M548	Carrier. Cargo. Tracked. 3442 kg(6-ton)	6V-53
M551	Armored Reconnaissance/Airborne	
	Assault Vehicle (Sheridan)	6V-53T
M561	Gamma Goat	3-53
M557A1	Carrier. Command Post. Light Tracked	6V-53
M578	Recovery Vehicle	8V-71T
M746	Heavy Equipment Transporter (Het 70)	12V-71T
XM667	Carrier. GM. Equipment. SP	a
XM727	Carrier. GM. Equipment. SP	а
XM730	Carrier. GM. Equipment. SP	a
XM741	Chassis, Gun, AA Artillery, 20mm, SP	a
XM806E1	Recovery Vehicle. FT Armored	а
	Truck, Dump, 18 140 kg (20-ton), Diesel	
	Electric Driven	6V-71

a = Vehicles are powered by either 6V-53, 6V-53T, or 8V-71T (TB-750-652).

recommends using diesel fuels with less than 0.5 wt% sulfur because "too high a sulfur content results in excessive cylinder wear due to acid build-up in the lubricating oil" (Ref 1). Previous investigations conducted by the United States Army Fuels and Lubricants Research Laboratory (USAFLRL) used an aluminum block engine model 6V-53T and revealed engine/fuel/lubricant incompatibilities when using fuels containing greater than 0.5 wt% sulfur and MIL-L-2104C (Ref 2) specification lubricants. The observed incompatibilities included catastrophic piston/ring/exhaust valve failure and relatively high deposit and wear rates (Ref 3). Additional documentation of the detrimental effects of high-sulfur diesel fuel can be found in References 4 through 11.

Diesel fuel specification VV-F-800B, OCONUS, (Ref 12) allows procurement of fuel with up to 0.7 wt% sulfur content. Based on this sulfur limit, and the previous USAFLRL test results with the two-cycle diesel engine and high-sulfur fuel, a program was initiated to identify methods of counteracting the detrimental effects of high-sulfur fuel. The program objective was to identify fuel and/or lubricant modifications which would allow continous operation on diesel fuel containing greater than 0.7 wt% sulfur without significantly reducing engine performance or service life. Identification of such fuel/lubricant modifications would expand the supply of diesel fuel available to the U.S. Army and potentially extend the service life of two-cycle diesel equipment. A previous report (AFLRL No. 105) covered the establishment of low- and high-sulfur fuel baselines using a constant lubricant in the iron block engine model 3-53 (Ref 13). The evaluation of various lubricants for their effectiveness in combating the effect of high-sulfur fuel was reported in AFLRL No. 109 (Ref 14) and AFLRL No. 127 (Ref 15). Lubricant effectiveness was defined in terms of how well the lubricant performed as compared to the low- and high-sulfur fuel baselines. Two of 12 lubricants tested (Ref 14 and Ref 15) had overall engine conditions, when using high-sulfur fuel (HSF), which approached the desired conditions observed when low-sulfur fuel was used. current report covers: (1) the initial evaluation of a fuel additive for its effectiveness in controlling the effects of using HSF, and (2) the further evaluation of a lubricant which had shown promise in counteracting high-sulfur fuel.

II. EVALUATION OF FUEL ADDITIVE (3-53 Test Number 22)

A. Test Engine (3-53)

An iron-block, two-cycle diesel engine Model 3-53 was utilized as the test engine. This engine is the powerplant used in the M561 1-1/4T tactical truck (Gamma Goat). Additionally, this engine was used to minimize test fuel and engine rebuild costs per test while still utilizing a "real-world" engine. Table 2 gives the characteristics of the 3-53 engine. The engine was fully instrumented and coupled to a laboratory test stand dynamometer as shown in Figure 1.

B. Test Technique

The test was conducted following the U.S. Army/CRC 210-hour wheeled-vehicle endurance cycle (Ref 16) which has been correlated to 32,200 km (20,000 miles) of proving ground operation. This test cycle includes alternating periods of full-power and cold idling with an overnight shutdown as shown in Table 3. A complete description of the detailed procedure is presented in Appendix A.

C. Approach

As reported in the literature (Ref 4 through 11), increasing diesel fuel sulfur content causes increased engine wear and deposition. These effects were quantified in the 3-53 engine by establishing a low-sulfur fuel baseline and a high-sulfur fuel baseline while using a constant lubricant (Ref 13). The low-sulfur fuel baseline serves as an example of the desired performance level. The overall program objective is to identify fuel and/or lubricant modifications which, when used with high-sulfur fuel, will result in engine

TABLE 2. 3-53 ENGINE CHARACTERISTICS

Engine Type	Normally Aspirated, Two-cycle compression
	ignition, direct injection, uniflow
	scavenoino

Weight (dry), kg (1b) 431 (950)

No. of cylinders, arrangement 3 in line

Displacement, liter (cu in.) 2.6 (159)

Bore and stroke, cm(in.) $9.84 \times 11.43 (3-7/8 \times 4-1/2)$

Cylinder block material cast iron (cast iron liners)

Rated power, kW(Hp) 72.3 (97) at 2800 rpm

Maximum torque, Nm(1b-ft) 278 (205) at 1800 rpm

Compression ratio 21 to 1

Fuel system Unit injector (N 50 needle valve),

primary and secondary engine filters

Governor Variable speed with throttle controls

Oil filter Full-flow single filter

Oil cooling Integral heat exchanger using 100 percent

jacket-coolant flow capacity - 13.2 1

(14 qts)

Piston description
Material/design Cast iron/trunk type

Ring configuration 1 - Fire ring (rectangular)

3 - Compression rings (rectangular)

2 - Oil rings

Piston cooling From jet in top of connecting rod

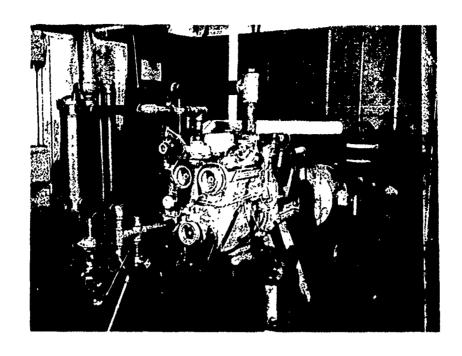


FIGURE 1. 3-53 TEST CELL INSTALLATION

DIESEL ENGINE MODEL 3-53 TEST FACILITY (FULL POWER FUEL CONS. = 6.3 GPH)

TABLE 3. U.S. ARMY/CRC 210-HOUR WHEELED VEHICLE ENDURANCE CYCLE

Period	Time, hr	Load, %	RPM	Coolant Temp, °C(°F)
1	2	100	2800	96(205)
2	1	0	650	38(100)
3	2	100	2800	96(205)
4	1	0	650	38(100)
5	2	100	2800	96 (205)
6	1	0	650	38(100)
7	2	100	2800	96 (205)
8	1	0	650	38 (100)
9	2	100	2800	96 (205)
10	10		Shutdown	· · · · · · · · · · · · · · · · · · ·
	$\frac{10}{24}$			

Complete test is 15 days at 14 hr/day for 210 hours.

condition similar to the low-sulfur baseline. The low-sulfur baseline was established using lubricant REO 203 and reference diesel fuel (0.4 wt% S) which is defined by Federal Test Method Standard 791B, Method 341.4. This combination had previously produced excellent results in the 6V53T engine (Ref The high-sulfur fuel baseline was established using diesel fuel containing 1.0 wt% sulfur and lubricant REO 203 (Ref 13).

Test Details

Test Number 22 was conducted in 3-53 engine number 3D131703, which was rebuilt with new cylinder kits (piston, rings and cylinder liner) and clean exhaust valves. Before test, the engine was measured for (1) liner bore (top/middle/hottom) at thrust/antithrust and front/back positions, (2) piston diameter, and (3) piston ring gap. Pre- and post-test full-load performance tests were determined using the test fuel.

The engine was operated in accordance with the procedure detailed in Appendix A and summarized in Table 3. The following hourly readings and calculations were made to monitor test operation:

> Engine Speed Engine Load Torque Observed Power Fuel Rate BMEP **BSFC**

Temperatures

Jacket Coolant-In Jacket Coolant-Out 011 Sump Inlet Air (Blower) Exhaust Manifold Fuel at Filter Fuel at Return

Pressures

Oil Gallery Blower Discharge Intake Vacuum Exhaust, Common Crankcase

Minimum and maximum values and averages of these readings and calculations are presented in Appendix B.

After each test, the engine was disassembled, and the following determinations were made:

Engine condition ratings in accordance with standard CRC methods (Ref 17, 18) for:

- Ring face burning 1.
- Ring sticking 2.
- Cylinder liner scuffing and glazing

- 4. Intake port deposits
- 5. Ring deposits
- 6. Piston deposits
- 7. Exhaust valve condition

Engine wear measurements for:

- Cylinder liner ID (top/middle/bottom)
- Ring gap
- 3. Piston diameter

Oil consumption was calculated, and photographs were made of various engine parts. Used oils were analyzed to determine chemical and physical property changes. The above items are all included in Appendix B.

E. Test Lubricant

The second secon

The engine oil used for the low-sulfur fuel baseline, high-sulfur fuel baseline, and Test Number 22 was REO 203. The properties of REO 203 are shown in Table 4.

TABLE 4. TEST LUBRICANT PROPERTIES (REO 203)

Property Jude	<u>Method</u>	Value REO 203
K Vis, cSt at 40°C K Vis, cSt at 100°C Viscosity Index TAN TBN	D 445 D 445 D 2270 D 664 D 2896 D 664	104.6 11.8 101 3.6 5.4 4.5
Insolubles, wt% Pentane A Benzene A Pentane B Benzene B	D 893	0.05 0.04 0.03 0.02
API Gravity, ° Pour Point, °C Flash Point, °C Carbon Residue Sulfated Ash, wt%	D 287 D 97 D 92 D 524 D 874	27.5 -21 241 1.19 0.93
Elemental, wt% Ba Ca Mg Zn P	Method AA AA (XRF) AA AA XRF	NIL 0.24 NIL 0.09 0.09

ND = Not determined.

3,2

XRF = X-ray fluorescence.

F. Discussion and Results

It has been reported in the literature that the addition of 0.3 yol% of an organo-zinc complex fuel additive (zinc naphthenate) to high-sulfur diesel fuel was an effective means of controlling corrosive engine wear (Ref 19). Test Number 22 was conducted to determine if the use of 0.3 yol% zinc naphthenate as a diesel fuel additive would be effective in controlling the deleterious effects observed when using high-sulfur diesel fuel in a two-cyc. diesel engine. A comparison is shown in Table 5 of the properties of the

TABLE 5. TEST FUEL PROPERTIES

Property	ASTM Method	LSF	B HSF	C HSF + Additive
API°	D 287	33.2	34.2	34.4
K. Vis, 40°C, cSt	D 445	3.20 ^a	2.87	2.98
Sulfur, wt%	D 1266	0.42	1.03	1.04
Copper Corrosion	D 130	1A	1A	1A
Carbon Residue	D 524	0.10	0.16	0.06
Cetane No.	D 613	47	52	52
Ash, wt%	D 482	0.006	0.001	0.035
Zn, wt%	XRF	N11	Nil	0.024
Distillation	D 86			
% off at °C				
IBP		210	208	202
19		242	238	235
50		271	269	268
90%		317	308	307
EP		365	349	346

a = Viscosity determined at 38°C.

following fuels used in this test program: (a) low-sulfur fuel, (b) high-sulfur fuel and (c) high-sulfur fuel plus additive. The low-sulfur fuel is defined by Federal Test Method 791B, Method 341.4. The high-sulfur fuel contained all straight run material and the sulfur content was 85 percent naturally occurring sulfur compounds with the balance of sulfur added as ditertiary butyl disulfide. Addition of 0.3 vol% zinc naphthenate to high-sulfur fuel increased the fuel ash to 0.035 wt% while the cetane number remained unchanged at 52.

Before discussing the results of Test Number 22, the key performance areas of the low- and high-sulfur fuel baselines will be reviewed. The results of the LSF baseline (Table 6) are representative of the desired engine condition at the end of the test. Measured wear (fire ring gap and cylinder liner bore) were low as were cylinder liner scuffing and ring face burning. Piston deposit levels were moderate, and no serious ring sticking problems (Table 7) were observed. Compared to the LSF baseline, the HSF baseline had much more severe ring face burning (32% versus 1%) and cylinder liner scuffing (41% vs 4%). Measured wear was two to four times more severe for the HSF baseline. Piston cleanliness had deteriorated slightly and ring sticking tendency had

TABLE 6. TEST RESULTS
Wear, Deposits, and Other Ratings

	Test Number		
	LSF	HSF	
	1	4, 12, 18	22
Wear Average Fire Ring Gap Change,			
μm	51	237	152
Average Cylinder Liner Fore Change,	_		
Front-Back and Thrust-Antithrust, μm	8	16	18
Thrust-Antithrust only, µm	8	23	25
Average Liner Scuffing, %	4	41	28
Average Liner Glazing, %	5	9	10
Deposition Piston WTD* Rating			
Cylinder l	226	393	424
Cylinder 2	318	374	255
Cylinder 3	356	345	305
Average	300	371	328
Average Port Restriction, %	7	1	8
Average Liner Lacquer, %	40	91	90
Other			
Average Ring Face Burning, % (Fire Ring + 1-3 Compression Rings)	1	32	28
Used Oil Iron Content, ppm			
at 210 hr by XRF	110	117	100

^{*}WTD = Weighted Total Deposit

TABLE 7. RING STICKING SUMMARY

Test No.	Ring Sticking (Cylinder-Ring-Condition)
l(LSF)	#2 - F/R - Sluggish #3 - F/R - 15% Cold Stuck
	#3 - F/R - 13% Cold Stuck
4(HSF)	#3 - F/R - Sluggish
12(HSF)	#2 - F/R - 60% Cold Stuck
18(HSF)	#1 - CR#2 - 5% Cold Stuck
22(HSF + Fuel Additive)	#2 - F/R - Sluggish

increased slightly, but still did not approach problem levels. As shown in Table 8, the used oil from the LSF test was still in satisfactory condition while the HSF lubricant had been degraded only slightly more (mainly an increase in flash point and a slight increase in viscosity). Our primary objective was to obtain engine condition approaching or equal to the LSF baseline condition when the engine was operated using HSF which had been modified by the addition of the fuel additive under evaluation.

Test Number 22 was run using REO 203 lubricant and high-sulfur fuel (1 wt% S) treated with 0.3 vol% zinc naphthenate. A targeted 210 hours were completed following the U.S. Army wheeled-vehicle test procedure. The average operating conditions for Test Number 22, the LSF, and HSF baselines are shown in Table 9. At the completion of Test Number 22, engine power had been reduced by about 10 percent, as shown in Figure 2. The reason for the observed power loss was traced to injector tip deposits which had accumulated during the test in all three cylinders as shown in Figure 3. Replacement of the fouled injectors with new injectors resulted in recovery of the power loss as shown in Figure 4. Also, the exhaust valves from Test Number 22 had incipient leakage as shown in Figure 5, which was probably caused by the increased fuel ash. This preliminary exhaust valve leakage had not become severe enough to reduce power output.

Post test engine wear measurements and deposit ratings for Test Number 22 are compared to the results of the LSF and HSF baselines in Tables 6 and 7. For Test Number 22, the average fire ring gap increase was reduced by about 35 percent compared to the HSF baseline, but did not approach the desired level observed for the LSF baseline. When using the zinc fuel additive, cylinder liner wear and ring face burning were not reduced, but average cylinder liner scuffing was reduced about 32 percent compared to the HSF baseline. The used oil properties for Test Number 22 are shown in Table 8. The primary difference in used oil properties between Test 22 and the HSF baseline was increased amount of coagulated insolubles (pentane and toluene) observed for Test 22.

Overall, the use of zinc naphthenate as a fuel additive to help control the deleterious effects of using high sulfur diesel fuel is not recommended. This is because of the fuel injector fouling observed during Test Number 22, incipient exhaust valve leakage and burning, and the overall engine condition which did not approach the LSF baseline.

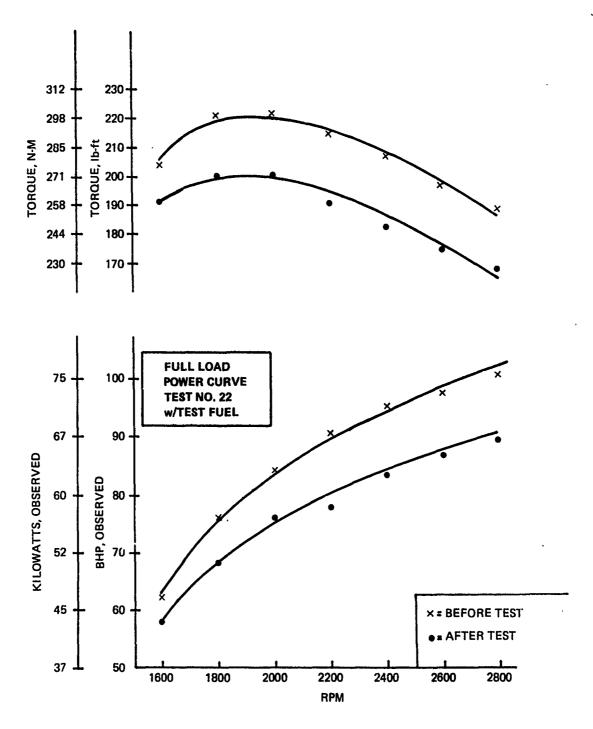
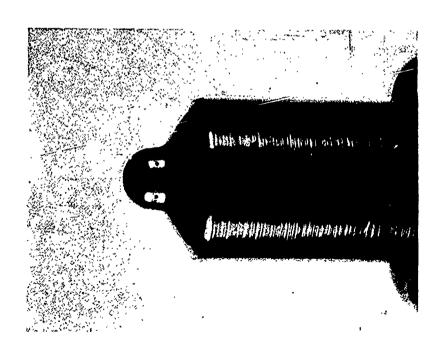


Figure 2. Full Load Power Curves Test No.22 With Test Fuel



Injector Tips

Figure 3.



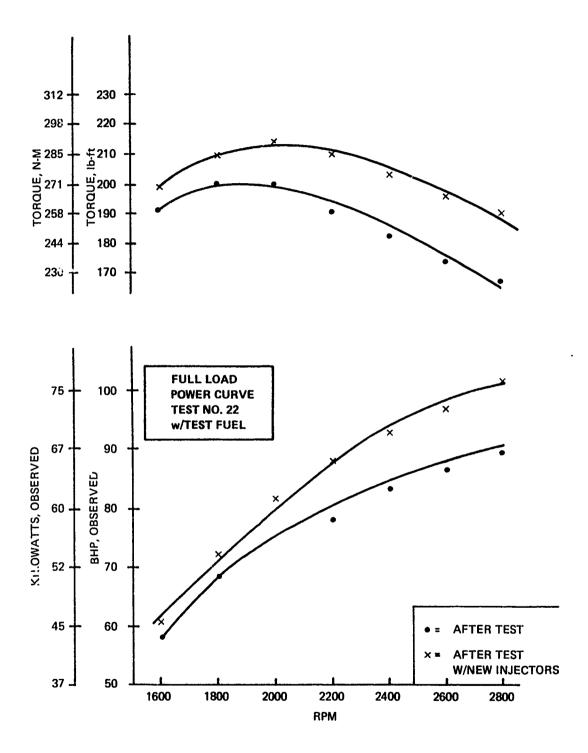
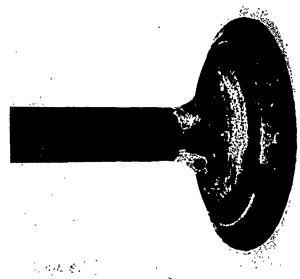


Figure 4. Effect of Injector Deposits on Full Load Power Curves



Typical Exhaust Valve Test #22

Figure 5.



(N)

TABLE 8. SUMMARY OF USED OIL ANALYSES

Property	Method	LSF Baseline	HSF Baseline	Test 22
K. Viscosity at 40°C, cSt at 210 hrs Δ from new	D 445	119.8 ^a -1.8	123.2 +18.6	121.2 +16.6
K. Viscosity at 100°C, cSt at 210 hrs Δ from new	D 445	12.9 ^b +0.3	13.4 +1.6	13.2 +1.4
TAN at 210 hrs Δ from new	D 664	3.5 -0.1	3.6 0.0	4.2 +0.6
TBN at 210 hrs Δ from new	D 2896	4.4 -1.0	3.6 -1.8	3.0 -2.4
Flash Point, °C at 210 hrs! Δ from new	D 97	238 -3	252 +11	238 -3
Carbon Residue, wt% at 210 hrs Δ from new	D 524	1.77 +0.58	2.07 +0.88	2.41 +1.22
Sulfated Ash, wt% at 210 hrs Δ from new	D 874	1.09 +0.16	1.19 +0.26	1.32 +0.39
Insolubles, wt%	D 893			
(with coagulent) Pentane at 210 hrs Δ from new		0.41 +0.38	0.40 +0.37	1.02 +0.99
Benzene (Toluene) at 210 hrs Δ from new		0.28 +0.26	0.12 +0.10	(0.92) (+0.90)
Elemental, ppm	AA (XRF)			
at 210 hrs Fe Cr Cu Pb		(110) ND (50) 2	82(117) 5 9 43	68(100) 5 7 4

a = Viscosity determined at 210°F. b = Viscosity determined at 100°F ND = Not Determined

TABLE 9. AVERAGE TEST OPERATING CONDITIONS

	Test		
Parameter	1(LSF)	4,12,18(HSF)	22
Power (observed), kW	71	72	70
Torque, nm	241	248	239
BMEP, kPa	586	598	572
Fuel Rate, kg/hr	19.6	19.2	19.5
BSFC, kg/kW-hr	0.276	0.264	0.278
Oil Temperature, °C	110	122	120
Exhaust Temperature, °C	507	523	52
Total Oil Consumption, kg	15.9	20.4	25.9

III. EVALUATION OF MIL-L-21260B LUBRICANT

A. Approach

A qualified lubricant which met the requirements of MIL-L-21260E, "Lubricating Oil, Internal Combustion Engine, Preservative and Break-In," (Ref 20) had previously been tested in the 3-53 engine using high-sulfur fuel (Ref 15). This test resulted in overall improvement in engine condition as compared to the HSF baseline for the 3-53 engine. Ring face burning and cylinder liner scuffing approached levels observed when using LSF. Fire ring gap increase, cylinder liner wear, and ring sticking were comparable to the HSF baseline. Because the use of this Preservative Engine 0il showed promise in controlling the deleterious effects of high sulfur fuel in the 3-53 engine, it was decided to test this lubricant using high-sulfur fuel in the higher output, turbocharged 6V-53T engine. In previous testing, the aluminum block version of this engine had been extremely sensitive to fuel and lubricant quality (Ref The results of the Preservative oil/high-sulfur fuel test (Test HSF-1) will be compared to a low-sulfur fuel baseline (LST/MIL-L-2104C Reference Oil, Test LSF-1), which was established in a different program (Ref 21). A comparison will also be made against a high-sulfur fuel baseline test (HSF/MIL-L-2104C Reference Oil, Test HSF-2) which was run as part of this program.

B. Test Engine (6V-53T)

A <u>cast iron-block</u>, two-cycle diesel engine model 6V-53T was utilized as the test engine. This engine is currently being retrofitted in the M551, Armored Reconnaissance/Airborne Assault Vehicle (Sheridan) as a replacement for the <u>aluminum-block</u> 6V-53T. Table 10 gives the characteristics of the 6V-53T engine. The engine was fully instrumented and coupled to a laboratory test stand dynamometer as shown in Figure 6.

C. Test Technique

The tests were conducted following the U.S. Army 240-hour tracked-vehicle test cycle which has been correlated to 6440 km (4000 miles) of proving ground operation (Ref 16). This test cycle includes alternating periods of operation at idle, maximum power, idle, and maximum torque as shown in Table 11. A complete description of the tracked-vehicle endurance test cycle is presented in Appendix C.

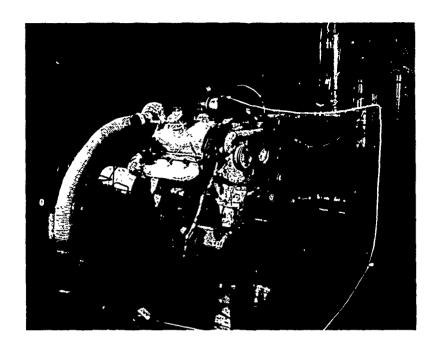


Figure 6. 6V-53T Test Cell Installation

TABLE 10. 6V-53T ENGINE SPECIFICATIONS

Engine Type	Turbocharged, Direct Injection, Uniflow Scavenged, Two-Cycle Compression Ignition
No. of cylinders, arrangement	6, V
Displacement, in. 3(liters)	318(5,2)
Bore x Stroke, in. (mm)	3.875 x 4.500(98.43 x 114.30)
Rated Power, Bhp(kW) Rated Torque, ft-1b(N-m) Oil Capacity, gal.(liters)	300(224) at 2800 rpm 615(834) at 2200 rpm 5(19)
Piston Material, Design	Cast Iron, Trunk-Type

TABLE 11. U.S. ARMY/CRC 240-HOUR TRACKED-VEHICLE ENDURANCE CYCLE

Period	Time, hrs	Rack Setting	Coolant Jacket-Out Temp, °F(°C)
1	0.5	idle	100(38)
	2.0	Maximum Power*	170(77)
	0.5	idle	100(38)
	2.0	Maximum Torque	170(77)
2	0.5	idle	100(38)
	2.0	Maximum Power	170(77)
	0.5	idle	100(38)
	2.0	Maximum Torque	170(77)
3	0.5	idle	100(38)
	2.0	Maximum Power	170(77)
	0.5	idle	100(38)
	2.0	Maximum Torque	170(77)
4	0.5	idle	100(38)
	2.0	Maximum Power	170(77)
	0.5	idle	100(38)
	2.0	Maximum Torque	170(77)
5	4	5 min idle, followed by shutdown	

These 5 periods yield 20 hours of running with a 4 hour shutdown; this cycle is repeated 12 times for a total test time of 240 hours.

^{*} For the 6V-53T, Maximum Power occurs at 2800 rpm and Maximum Torque occurs at 2200 rpm.

D. Test Details

The tests using high-sulfur fuel were conducted in 6V-53T engine number 6D-178671 which was rebuilt for each test with new cylinder kits (piston, rings, and cylinder liner), new piston pins and clean exhaust valves. A summary of the applicable 6V-53T engine tests is shown in Table 12. An Appendix which contains the following items is included for each HSF test:

TABLE 12. SUMMARY OF 6V-53T TESTS

Test No.	Fuel	Lubricant	Comment
LSF-1	0.4 wt% S	Army Reference 011 - MIL-L-2104C	LSF baseline (Ref-2:)
HSF-1	1.0 wt% S	MIL-L-21260B	Test lubricant (AL-9065)
HSF-2	1.0 wt% S	Army Reference 0i1 - MIL-L-2104C	HSF baseline

- Complete engine rebuild measurements
- Mean and standard deviation of the operating conditions
- Pre- and post-test power curves
- New and used lubricant analyses
- Wear measurements
 - Cylinder liners
 - Piston rings
- Engine condition observations and deposit ratings using standard CRC methods (Ref 17)
- 0il consumption data
- Photographs
 - Faces of king Sets
 - Best and worst cylinder liner and corresponding piston

E. Discussion and Results

The properties of the qualified MIL-L-21260B Preservative test oil (AL-9065) and the U.S. Army MIL-L-2104C reference engine oil (AL-8980) are shown in Table 13. The preservative oil differs from a MIL-L-2104C lubricant in additive content. A supplemental zinc-containing additive is added to a MIL-L-2104C oil to produce the MIL-L-21260B Preservative Oil.

Table 14 contains the properties of the low- and high-sulfur test fuels. Low-sulfur fuel (LSF) is a reference DF-2 which is defined by Federal Test Method Standard 791B, Method 341.4. The high-sulfur fuels (AL-9242 and AL-9697) were two separate acquisitions of a single batch of high-sulfur fuel prepared by the Howell Corporation, San Antonio, Texas. The HSF contained all straight run material and all of the sulfur content was naturally occurring sulfur compounds.

TABLE 13. TEST LUBRICANT PROPERTIES

Property	Test Method	AL-8980 Ref 011 A MIL-L-2104C	AL-9065 Preservative Oil MIL-L-21260B
K. Viscosity at 40°C, cSt K. Viscosity at 100°C, cSt Viscosity Index Total Acid No., mg KOH/g Total Base No., mg KOH/g Total Base No., mg KOH/g Flash Point, °C API Gravity at 16°C(60°F)	D 445 D 445 D 2270 D 664 D 664 D 2896 D 92 D 287	109.1 11.7 93 2.3 13.3 13.6 223 25.5	106.8 11.7 97 2.2 9.4 10.6 246 27.8
Carbon Residue, wt% Sulfated Ash, wt% Elements, wt% Ca Zn P S	D 524 D 874 AA AA XRF XRF	2.10 1.78 0.48 0.07 0.07 0.65	1.28 1.41 0.36 0.15 0.08 0.23

TABLE 14. TEST FUEL PROPERTIES

Property	Test <u>Method</u>	LSF	HSF (AL-9242/AL-9697)
API° K. Viscosity at 40°C, cSt Sulfur, wt% Copper Corrosion Carbon Residue Distillation	D 287 D 445 D 1266 D 130 D 524 D 86	33.2 3.20 ^a 0.42 1A 0.10	31.5 3.02 1.02 1A 0.08
% off at °C IBP 10% 50% 90% EP Gross Heat of Combustion, mj/kg	D 240	210 242 271 317 365	214 243 278 317 338

a = Viscosity determined at 38°C.

The state of the s

Table 15 contains the key average operating conditions for these three 6V-53T engine tests. Parameters included are: power, torque, fuel consumption, and

TABLE 15. AVERAGE OPERATING CONDITIONS

Test No.	LSF	-1	HSF	-1	HSF	-2
	2200	2800	2200	2800	2200	2800
Engine Speed	rpm	rpm	rpm	rpm	rpm	rpm
Power, kW	185	224	201	231	<u>rpm</u> 203	228
Torque, N-m	802	761	872	793	879	776
Fuel Rate, kg/hr	44.2	54.8	48.6	56.5	48.7	55.2
BSFC, kg/kW-hr	0.239	0.245	0.241	0.244	0.240	0.243
Oil Sump Tempera ure,						
•C	119	121	113	118	109	114
Fihaust Temperature,						
°C (preturbo)	529	544	561	542	556	532

operating temperatures. Table 16 contains tests results including wear measurements for fire ring gap and cylinder liner bore, piston and liner deposition ratings, and other important ratings. Table 17 shows a tabulation of ring sticking performance for each test. Finally, Table 18 contains the used oils analyses at end of test (240 hours) and the change in property value from new. Having presented an overall summary of test operation and results, each high-sulfur fuel test (HSF-1 and HSF-2) will be discussed individually.

Test HSF-1 was run using MIL-L-21260B Preservative engine oil (AL-9065) and high-sulfur fuel. The targeted 240 hours were completed without problems. At the completion of HSF-1, only Test LSF-1 was available for comparison. evaluating the effectiveness of AL-9065 in combating the effects of highsulfur fuel, the primary objective was to obtain engine condition approaching or equal to the baseline condition Test LSF-1, when the engine was operated using high-sulfur fuel. As shown in Table 16, the use of the preservative oil with high-sulfur fuel resulted in slightly less measured liner wear, and liner scuffing, and reduced ring face burning compared to Test LSF-1. Test HSF-1 had about three times the fire ring gap increase as Test LSF-1. No ring sticking was observed for Test HSF-1, and the used oil properties (Table 18) revealed that AL-9065 was only slightly degraded. The TBN had been reduced by 53 percent, but all other properties were well within the oil change limits stated by the engine manufacturer (Ref 1). It should be noted that at 240 hours (end of test), the lubricant has been in service only 120 hours as the test procedure requires an oil change at 120 hours. Overall, the use of preservative oil resulted in engine condition which was equivalent to or better than the LSF baseline, with the exception of fire ring gap increase. The next step in this program was to test the MIL-L-2104C reference oil with high-sulfur fuel to provide a HSF baseline.

Test HSF-2 was run using MIL-L-2104C reference oil (AL-8980) and high-sulfur fuel. The targeted 240 hours were completed, but several operational problems were encountered which made the integrity of this test questionable. At 25.5 hours, an oil sight glass, which is used to monitor engine oil level during shutdowns, broke. About 42 percent of the engine oil was lost, and subsequently replaced with fresh lubricant. At 40 hours, the splines on the shaft

TABLE 16. SUMMARY OF 6V-53T TEST RESULTS

(MIL-L-2104C) (MIL-L-2104C) Test Hours 240 240 240 240 Wear Avg Fire-ring gap increase, μ m 76 254 114 Avg Cylinder Liner Bore increase, μ m 76 23 23 23 All 30 18 18 Avg Cylinder Liner Scuffing, % 17 10 11 Avg Cylinder Liner Glazing, % 21 17 12 Deposition Piston, WTD 1 359 363 390 2L 369 375 420 3L 298 400 394 1R 384 420 319	Test No. Fuel Lubricant	LSF-1 0.4 wt% S Ref Oil A	HSF-1 1.0 wt% S MIL-L-21260B	HSF-2 1.0 wt% S Ref Oil A
Avg Fire-ring gap increase, μ m 76 254 114 Avg Cylinder Liner Bore increase, μ m Thrust-Antithrust only 36 23 23 All 30 18 18 Avg Cylinder Liner Scuffing, % 17 10 ^a 11 Avg Cylinder Liner Glazing, % 21 17 12 Deposition Piston, WTD 1L 359 363 390 2L 369 375 420 3L 298 400 394	Test Hours	(MIL-L-2104C) 240	240	(MIL-L-2104C) 240
Thrust-Antithrust only 36 23 23 All 30 18 18 Avg Cylinder Liner Scuffing, % 17 10 ^a 11 Avg Cylinder Liner Glazing, % 21 17 12 Deposition Piston, WTD 1L 359 363 390 2L 369 375 420 3L 298 400 394	Avg Fire-ring gap increase, µ m	76	254	114
All Avg Cylinder Liner Scuffing, % 17 10 ^a 11 Avg Cylinder Liner Glazing, % 21 17 12 Deposition Piston, WTD 1L 359 363 390 2L 369 375 420 3L 298 400 394		36	23	23
Avg Cylinder Liner Scuffing, % 17 10 ^a 11 Avg Cylinder Liner Glazing, % 21 17 12 Deposition Piston, WTD 1L 359 363 390 2L 369 375 420 3L 298 400 394	•			
Avg Cylinder Liner Glazing, % 21 17 12 Deposition Piston, WTD 359 363 390 2L 369 375 420 328 400 394			10 ^a	
Deposition Piston, WTD 1L 359 363 390 2L 369 375 420 3L 298 400 394				
2R 372 403 323 3R 430 353 342 Avg of all 369 386 365 Avg Port Restriction, % 4 2 3 Compression Rings, Groove Carbon, 27 33 26 Piston Groove Inside Diameter, avg % 21 36 22	Piston, WTD 1L 2L 3L 1R 2R 3R Avg of all Avg Port Restriction, % Compression Rings, Groove Carbon, avg %, fill Piston Groove Inside Diameter, avg %	369 298 384 372 430 369 4	375 400 420 403 353 386 2	420 394 319 323 <u>342</u> 365 3
Other Avg Ring Face Burning, % Fire-ring 43 19 8 All Compression Rings 34 15 30 Exhaust Valves, Condition Lt. pitting & Lt. Leaking Lt. pitting & Lt. leaking Fe, used oil, ppm, XRF at 120 hr 74 84 85	Avg Ring Face Burning, % Fire-ring All Compression Rings Exhaust Valves, Condition	34 Lt. pitting & Lt. leaking	15 Lt. Leaking	30 Lt. pitting & Lt. leaking
Fe, used oil, ppm, XRF at 120 hr 74 84 85 at 240 hr 88 66 46			= *	

a = Had large amount of "light vertical lines," not counted as scuffing.

TABLE 17. RING STICKING SUMMARY

Test No.	Ring Sticking (Cylinder-Ring-Condition)
LSF-1	2L - Fire Ring - 5% Cold Stuck 3R - Fire Ring - 40% Cold Stuck 1R - Ring No. 2 - Collapsed
HSF-1	None
HSF-2	2L - Fire Ring - Partially Collapsed

TABLE 18. SUMMARY OF USED OIL ANALYSES (AT END OF TEST)

	Test		240 Hours	
Property	Method	LSF1	HSF-1	HSF-2
K. Viscosity at 40°C, cSt	D 445			
at 240 hrs		149.6	117.3	138.6
Δ from new		+42.9	+10.5	+31.9
K. Viscosity at 100°C, cSt	D 445			
at 240 hrs		13.4	12.6	13.6
Δ from new		+1.7	+0.9	+1.9
Total Acid No. at 240 hrs	D 664	3.5	2.6	3.0
Δ from new		+1.2	+0.4	+0.7
Total Base No. at 240 hrs	D 2896	12.0	(4.4)	(5.5)
∆ from new	(D 664)	-1.9	-5.0	-8.4
Flash Point, °C at 240 hrs	D 97	240	244	218
Δ from new		+17	-2	 5
Carbon Residue, wt% at 240 hrs	D 524	3.32	1.67	2.44
∆ from new		+1.22	+0.39	+0.34
Sulfated Ash, wt% at 240 hrs	D 874	2.01	1.58	2.01
Δ from new		+0.23	+0.17	+0.23
Insolubles, wt% at 240 hrs	D 893			•
(with coagulent) Pentane		1.14	0.20	0.19
Toluene		0.88	0.18	0.15
	XRF	0,00	0.10	0.13
Wear Metals, ppm at 240 hrs	AAF	88	66	46
Fe		ni1	1111	10
Cu		MTT	HTT	10

which drives the engine blower stripped. The shaft was replaced and the engine oil was changed as a precaution. At 157.5 hours, a leaking fuel injection line was discovered and repaired, and the engine oil was changed due to fuel dilution. Several used oil samples which were taken prior to the injector line repair were analyzed for fuel dilution using a gas chromatographic technique. It was found that fuel dilution had ranged from 20 to 35 vol% over the first 150.5 hours. The fuel dilution reduced engine oil viscosity, but viscosity had remained within the SAE 20 grade. Finally, the fuel filters had to be replaced four times during the test due to clogging and a resultant reduction in fuel flow and power. No fuel filter clogging was observed during Test HSF-1, which used an earlier sample of this same batch of HSF. Between Tests HSF-1 and HSF-2, the fuel had become unstable as evidenced by a gum content of 370 mg/100 ml.

A shown in Table 16, Test HSF-2 had overall engine condition which was very similar to Test HSF-1 and Test LSF-1. Fire ring gap increase was about 1.5 times the level observed in Test LSF-1, and fire ring face burning was substantially reduced in Test HSF-2. The used oil analyses (Table 18) show that the TBN had been reduced 63 percent, while other used oil properties remained within the oil change limits of the engine manufacturer (Ref 1). It is of interest to note that the end of test engine condition gave no indication that the engine had been operated with fuel-diluted lubricant. It is also postulated that the frequent oil changes, which were necessitated by mechanical failures, contributed to the unexpected good engine condition at end of test.

IV. CONCLUSIONS/RECOMMENDATIONS

The following conclusions are made based on the work reported herein:

A. Fuel Additive Evaluation

The use of zinc naphthenate as a fuel additive showed some potential for controlling fire-ring wear when using high-sulfur fuel; however, it is not recommended for use because of the following problems:

- (1) Power loss due to fuel injector fouling occurred with this additive.
- (2) The increased fuel ash content was contributing to exhaust valve leakage, and exhaust valve burning appeared to be imminent.
- (3) Overall engine condition did not approach the desired level observed with low-sulfur fuel.
- B. Lubricant MIL-L-21260B Evaluation
- (1) Except for fire-ring wear, the MIL-L-21260B/high-sulfur fuel combination resulted in engine condition which was very similar to that of the low-sulfur fuel baseline (Reference MIL-L-2104C oil/low-sulfur fuel).
- (2) Fire-ring wear was about 3.3 times more severe than the low-sulfur fuel baseline when the MIL-L-21260B lubricant and high-sulfur fuel were used.
- (3) Results of the reference MIL-L-2104C oil/high-sulfur fuel combination (HSF-2) are inconclusive because of the operational-mechanical problems encountered during this test.

- (4) Engine condition for Test HSF-2 was much better than expected. The more frequent oil changes, which were necessitated by the mechanical failures, probably helped to counteract the effects of using high-sulfur fuel.
- (5) The cast iron block 6V-53T engine appears to be much less sensitive to fuel sulfur content than the aluminum block 6V-53T engine.

The following recommendations for future effort are offered:

- (1) A controlled field test should be conducted using this MIL-L-21260B Preservative Engine Oil and high-sulfur fuel. The MIL-L-21260B lubricant gave engine condition about equivalent to the LSF baseline except for fire-ring wear. In our judgment, this lubricant is still the best overall candidate for further testing.
- (2) Research aimed at understanding the basic high-sulfur fuel combustion and engine degradation mechanisms is needed. With this basic HSF information, fuel or lubricant additives could be synthesized to specifically counteract the sulfur-related engine degradation mechanisms.
- (3) Future Army diesel engine oil specifications should include both two- and four-cycle diesel engine test requirements using high-sulfur fuel.

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Assistant Editor
Engine Deposit Ratings
Photography
Typing

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APPENDIX A

U.S. ARMY WHEELED-VEHICLE TEST PROCEDURE

U.S. ARMY WHEELED-VEHICLE TEST PROCEDURE DD 3-53 ENGINE

Test No.	: 3-53-703-	Engine Ser.	No.:	3D-131703	Test Cell	No.:	2
Test Lub	ricant			Fuel:			

Instructions

- 1. Pre-Test Preparations.
- 1.1 <u>Filter Elements.</u> Install new element in oil filter and change oil in air filter bath (using test oil).
- 1.2 Sump Oil Charge. Charge engine sump to full mark on dipstick with test oil (AL- -L). Close filler cap and motor engine for one minute at low speed (about 500 rpm) to fill oil cooler, filter, and internal oil rassages. Recheck level and add to full mark again (should be about 25 lb).
- 1.3 <u>Primary Fuel System.</u> After changing over to IH Cat fuel and flushing fuel lines, remove the Allen plug from top of primary fuel filter and fill the filter with fuel, then re-install plug.
- 1.4 Break-In Procedure. Set jacket coolant-out temperature controller at 205°F. Start engine and idle at 650 rpm for five minutes, then warm up at about 1000 to 1200 rpm for ten minutes. If no engine malfunctions or leakages occur, conduct the following break-in and record complete log sheet readings at end of each setting. Calculate: BHP, Torque, BSFC, BMEP.

	Load,	
Speed, rpm	1b-ft Torque	Jacket-Out, °F
1800	25	205
2200	55	205
2500	80	205
2800	80	205
	1800 2200 2500	Speed, rpm 1b-ft Torque 1800 25 2200 55 2500 80

PRECEDING PAGE BLANK-NOT FILLED

1.5 Full Load Performance Test. Following the break-in run, conduct a full load performance test run at the following conditions. Allow conditions to stabilize at each speed, then record complete log sheet readings at end of each setting. Calculate BHP, Torque, BSFC, BMEP.

Speed, rpm	Jacket-Out, °F
1600	205
1800	205
2000	205
2200	205
2400	205
2600	205
2800	205

- 1.6 <u>Valve Clearance Check.</u> Upon completing the full load performance test, stop engine and <u>immediately</u> check the <u>hot</u> clearance of the exhaust valves. Adjust clearances to 0.023-0.025, also injector height per gauge.
- 1.7 Oil and Fuel Change-Over. Completing valve clearance check, drain oil sump and filter. Discard drain and oil filter element. Weigh, record (on oil consumption log) and install new oil filter element, then charge system with full charge of test oil (AL- -L) as in item 1.2. Record weight of total charge. Change over to test fuel (AL- -F Tank #12) and flush fuel lines. Replace fuel filter elements (2) and prime as in item 1.3. Weigh oil blower can and record (oil consumption log).
- 1.8 <u>Full Load Performance Test</u>. Following fuel change-over, run full load performance test as in item 1.5.
- 2. Test.
- 2.1 Warm-Up. At the start of each day--idle for five minutes, then start test cycle 2800 rpm.
- 2.2 Test Conditions. After warm-up, set the following test conditions:

Test Cycle for 15 Days

Period	Time, Hrs	Load, %	Rpm	Temp., °F
1	2	100	2800±20	205±2
2	1	0	650±25	100±2
3	2	100	2800	205
4	1	0	650	100
5	2	100	2800	205
6	1	0	650	100
7	2	100	2800	205
8	1	0	650	100
9	2	100	2800	205
10	10		Shut Down	

Operate at test conditions 14 hours/day for a total of 210 hours. Complete log sheet readings at end of each period. Calculate: BHP, Torque, BSFC, BMEP.

- 2.3 <u>Daily Cool-Down</u>. After the last test hour each day, reduce the speed to idle (600-650 rpm) for five minutes (without resetting coolant controller), then stop engine.
- 2.4 <u>Used Oil Samples</u>. Flush oil filter tap, and withdraw a used oil sample during daily 5-minute cool-down (item 2.3) according to the <u>Oil Consumption</u> Log schedule and record sample weight.

Identify each sample as to test hours, test no., and oil code (AL- -L). Take: 2 oz. sample each day except at 70 and 140 hours take 12 oz. sample. At end of test, take 16 oz. sample. Take daily samples to Chem Lab for elemental analyses by XRF.

- 2.5 Oil Additions. New test oil additions, if required, are to be made at the end of each day after shutdown. Allow five minutes for oil to drain back to sump. Add weighed new oil to restore sump level to full by dipstick. Record weight of add-on oil consumption log.
- 2.6 <u>Final Oil Drain</u>. At end of after test power curve while engine is warm, drain sump, saving one gallon of used oil in clean can. Tag can, showing test No., oil code, date, and test hour. Also remove ilter element, weigh, and record.

2.7 Notes and Limits.

- (1) Coolant is 50% glycol/50% water.
- (2) Temperature must be reduced to 100°F within 15 minutes after idle starts.
- (3) Limits: Coolant Temp. ±2°F

 Oil Sump Temp. 265°F max

 Fuel at Filter Temp. 90°±5°F (105°F max=shutdown)
- (4) No Oil Change During Test.
- 3. After Test.

- 3.1 Full Load Performance Test. At end of test, run full load performance test as in item 1.5.
- 3.2 <u>Valve Clearance Check</u>. Upon completing end of test power curve, item 3.1, check hot valve clearances and record.
- 3.3 Wear and Deposits. Upon disassemble of engine, check wear measurements and deposit ratings (on sheets provided).
- 3.4 Record amount of fuel used for test.
- 3.5 <u>Calculations</u>. BHP (obs.) = Load x rpm/3000

 Torque (1b-ft) = Load x 1.75

 BSFC (1b/Bhp-hr) = 1b fuel per hr/Bhp (obs.)

 BMEP (psi) = Torque x 0.474
- 4. Cell Notebook.
- 4.1 Keep cell notebook updated (like a diary) at all times. Record what is being done (changes or repairs) to the cell, engine, instruments, etc. Record anything unusual and all modifications.

APPENDIX B

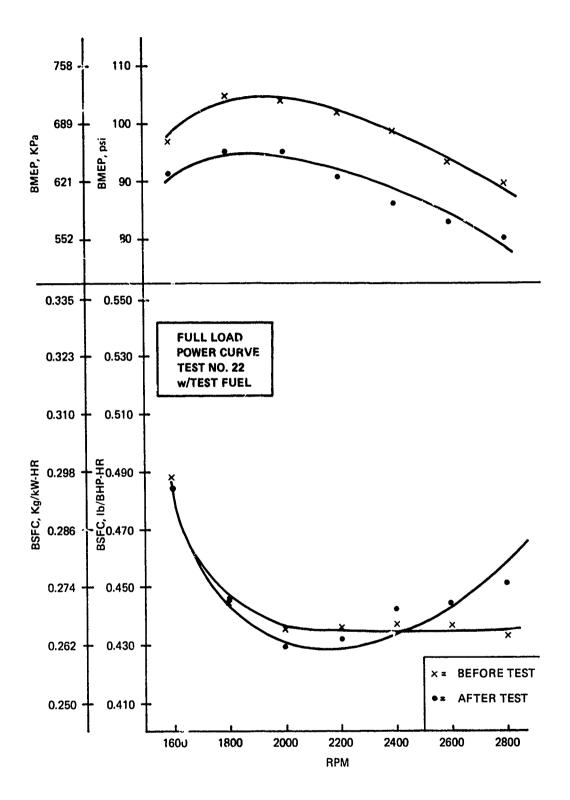
3-53 Test #22
Fuel: AL-8818-F (HSF + Zn)
Lube: AL-8822-L, REO 203
Start: 22 October 1979
End: 9 November 1979

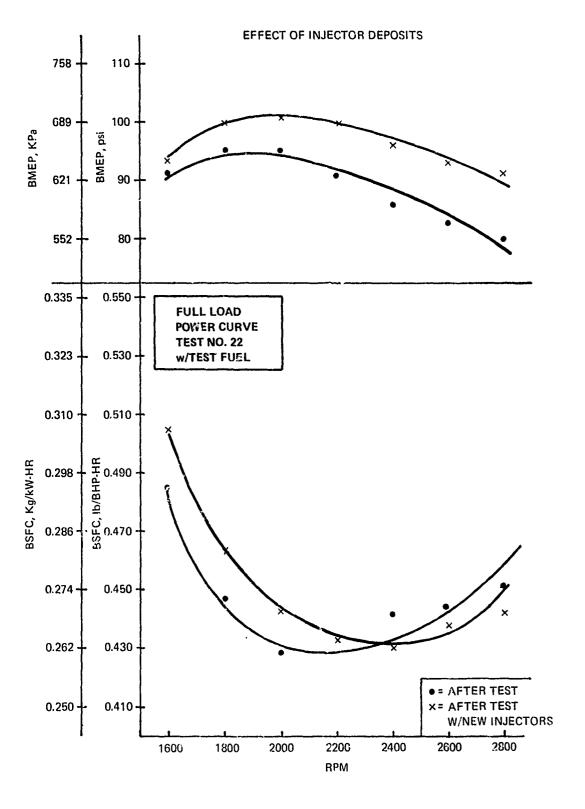
ENGINE OPERATING DATA (AVG) TEST #22

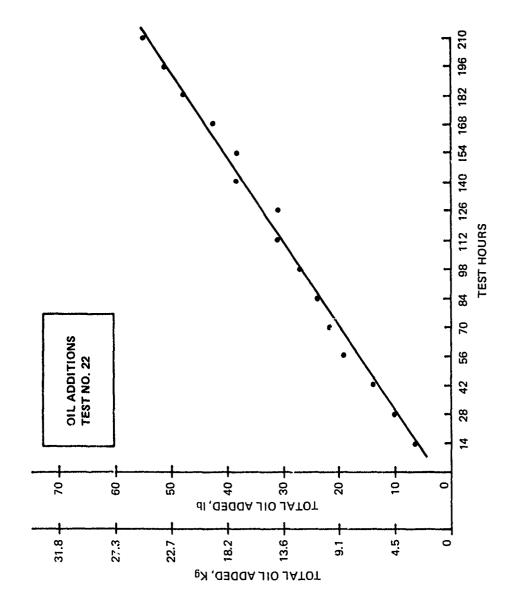
		Power		Idle
	Min	Max	Avg	(Avg)
Engine Speed, rpm	2791	2816	2801	641
Load, 1bs	97	107	101	
Torque, 1b-ft	170	187	176	
BHp obs	90	100	94	
Fuel Rate, 1b/hr	39.3	44.5	42.9	
BMEP, psi	81	89	83	
BSFC 1b/BHp-hr	0.434	0.480	0.457	
Temperatures, °F				_
Jacket Coolant-In	190	204	198	97
Jacket Coolant-Out	198	212	203	98
Oil Sump	242	258	248	
Inlet Air (Blower)	68	90 ·	79	
Exhaust Manifold	920	1020	990	
Fuel @ Return	139	156	145	
Fuel @ Filter	86	100	92	
Pressures				
Oil Gallery, psig	41	45	44	
Blower Discharge, psig	3.6	4.8	4.6	
Intake Vacuum, in. H ₂ 0	6.2	6.8	6.6	
Crankcase, in. H ₂ 0 2	0.28	0.37	0.36	
Exhaust, Common, in. Hg	0.4	2.7	2.3	
Transfer Pump, psig	74	75	74	
Oil Consumption, 1b.			57	

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LUBRICANT ANALYSES TEST #22

		New	70	140	210
D	Method	011	Hrs	Hrs	Hrs
Property	D445	104.6	111.98	119.03	121.18
K. Vis, cS, 40°C	D445	11.8	12.49	13.05	13.15
K. Vis, cS, 100°C	D2270	101	103	103	102
VI	D664	3.6	3.4	2.7	4.2
TAN	D2896	5.4	3.3	2.8	3.0
TBN	D893	,			
Insolubles, wt%	0073	0.05	ND	ND	0.05
Pentane A		0.04	ND	ND	0.05
Toluene A		0.03	ND	ND	1.02
Pentane B		0.02	ND	ND	0.92
Toluene B	D207	27.5	ND	ND	26.9
API Gravity,	D287	241	ND	ND	238
Flash Point, °C	D92	1.19	1.89	2.31	2.41
Carbon Residue, wt%	D524	0.93	1.15	1.28	1.32
Sulfated Ash, wt%	D874	0.93	1.13	1000	
Elemental	Method	0.24	ND	ND	ND
Ca, wt%	AA		0.17	0.20	0.20
Zn, wt%	AA	0.09	ND	ND	7
Cu, ppm	AA	ND	ND ND	ND	5
Cr, ppm	AA	ND		ND	4
Pb, ppm	AA	ND	ND 20/26	88/61	100/68
Fe, ppm	XRF/AA	ND	39/36	00/01	100,00

ND = Not Determined
AA = Atomic Absorption
XRF = X-Ray Fluorescence

PISTON SURFACE CONDITION TEST #22

	Piston Number				
	1	2	3		
Top Land	Norma1	Normal	Normal		
Skirt	Lt Scuff & Scratches	Lt Scratches	Lt Scratches		
Piston Pin	Normal	Normal	Normal		

PISTON GROOVE INSIDE DIAMETER - Z RING SUPPORTING CARBON TEST #22

		P:	iston Numbe	r
Piston Ring	Quadrant	1	2	3
1	1 2 3 4	95 0 0 15	0 0 0	5 5 10 5
2	1 2 3 4	0 100 100 85	15 0 75 0	5 0 0 90

 $\frac{Quadrants:}{1 = Thrust}$

2 = Rear

3 = Anti-thrust

4 = Front

RING DEPOSITS TEST #22

CARB 100-8	100-9	100-8	100–5	0	0	0	100-9	5-9 85-3 10-7	100-3	100-7	100-4
3 CARB 0	0	0	0	100 - ½ AHC	100 - AHC	100 - ½ AHC	0	0	0	0	0
LACQ	10-8	90-3 100-5	5-8 95-4	0	0	0	100-9	100-2	100-3	100-6	100-3
2 CARB 100 - ½ AHC	0	O	0	100-AHC	100 - ½ AHC	100 - ½ AHC	0	0	0	0	0
10-7	90-6 100-8	100-4	10-7 90-3	0	0	25–9	0	100-2	10-6	90~2 100~3	100-8
CARB	0	0	0	100-½AHC	25-B	85AHC 75-1-AHC	100-1,AHC	0	0	0	0
Cylinder Number Ring Top	2	٣	7	H	7	က	4		2	ო	7
Cylinder	,			ΙD		45		Bottom		•	

RING FACE CONDITION: % BURNING TEST #22

		Cylinder Number	
	1	2	3
First Ring	15	N	15
Second Ring	45	15	20
Third Ring	35	35	35
Fourth Ring	85	10	30
Average of all	28		

N = Normal

Some rings are slightly discolored, but not burnt.

RING STICKING TEST #22

Ring	Piston Number				
Ring No.	1	22	3		
1	F	Sluggish	F		
2	F	F	F		
3	F	F	F		
4	F	F	F		

F = Free

CYLINDER LINERS TEST #22

Cylinder Liner Scuffing Percent of Compression Ring Travel Area

			Travel Are	ea		
Cylind	er Percent Port	Perce	nt Scuffed	% Total		
Numbe	r Restriction	Thrust	Anti-Thrust	Area Scuffed	% Glazed	% Lacquer
1	10	20	75	47	10	90
2	2	10	15	13	10	90
3	5	15	30	22	10	90
Averag	ge 8	15	40	28	10	90

PISTON O.D. (IN) TEST #22

Cylinder	1	2	3
Before	3.8718	3.8712	3.8700
After	3.8715	3.8710	3.8701
Change	-0.0003	-0.0002	+0.0001

EXHAUST VALVE DEPOSITS TEST #22

	Cylinder No.	
Area	1 2	3
Head		
Face		
Tulip		
Stem		

EXHAUST VALVE SURFACE CONDITIONS TEST #22

	Cylinder No.				
Freeness in Guide	1 2 F	3 F			
Head	N N	N			
Face	Slight Leakage - due to	deposits			
Seat	Slight Leakage - due to	deposits			
Stem	n n	N			
Tip	n n	N			
F = Free N = Normal					

CYLINDER LINER I.D. (IN) TEST #22

			Front/Back	ς	Thre	ust/Antith	rust
Су	linder	Para	allel to (Crank	Perpen	dicular to	Crank
•	No.	Top	Middle	Bottom	Top	Middle	Bottom
1.	After	3.8758	3.8757	3.8761	3.8768	3.8766	3.8765
	Before	3.8750	3.8752	3.8755	3.8751	3.8754	3.8758
	Change	0.0008	0.0005	0.0006	0.0017	0.0012	0.0007
2.	After	3.8759	3.8755	3.8763	3.8760	3.8766	3.8766
	Before	3.8753	3.8754	3.8757	3.8755	3.8756	3.8 759
	Change	0.0006	0.0001	0.0006	0.0005	0.0010	0.0007
3.	After	3.8758	3.8757	3.8760	3.8767	3.8762	3.8763
	Before	3.8754	3.8754	3.8755	3.8752	3.8753	3.8755
	Change	0.0004	0.0003	0.0005	0.0015	0.0009	0.0008
Ave	rage (All)	0.0007	In				
	rage T/AT	0.0010	Ln				

PISTON RING GAP (IN) TEST #22

			R	ing No.	
Pis	ton No.	1	2	3	4
1.	After	0.043	0.028	0.033	0.028
	Before	0.037	0.028	0.033	0.027
	Change	0.006	0.000	0.000	0.001
2.	After	0.034	0.029	0.029	0.027
	Before	0.030	0.029	0.028	0.026
	Change	0.004	0.000	0.001	0.001
3.	After	0.045	0.027	0.026	0.028
	Before	0.037	0.026	0.025	0.027
	Change	0.008	0.001	0.001	0.001

Avg F/R (#1) Wear 0.006 In.

DAILY IRON WEAR METAL BY XRF TEST #22

Test	Iron
Hours	ppm
14	n11
28	32
42	40
56	44
70	39
84	54
98	66
112	67
126	79
140	88
154	82
168	91
182	96
	103
196	100
210	100

CRC DIESEL RATING SYSTEM

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STANDARD COMPUTATION SHEET FOR PISTON RATING LABORATORY TEST NUMBER 703-22 STAND NO. 2 ENGINE NO. 3D-131703 FUEL AL-8818-F RATER LYONS TEST LABORATORY AFIRE LUBRICANT AL-8822-L TEST HOURS 210 TEST PROCEDURE_

PISTON NO. _

DATE 11-14-79

NO. 1 GROOVE, VOLUME-%

															L					
																PISTO	WTD.	PISTON WTD. RATING	ပ္ခ	305
						GROO	OVES							LA	LANDS				3	DER.
٥	DEPOSIT	DEPOSIT	Ö	1.3	N _O	NO. 2	ž	NO. 3	ž	NO. 4	NO.	. 1	NO. 2	.2	N	NO. 3	NO.	5.4	g S	CROWN
	٠ ۲	FACTOR	AREA.%	AREA.% DEMENIT AREA.% DEMERI	AREA-%	_	AREA.%	DEMERIT	AREA.%	DEMERIT	AREAK	AREAS DEMERIT AREAS DEMERITAREAS DEMERITAREAS DEMERITAREAS DEMERITAREAS DEMERITAREAS DEMERITAREAS DEMERIT	NREA-%	DEMERIT	AREA-%	DEMERIT	AREA.%	DEMERIT	AREA.%	DEME
	2 H	8.	10	10.00	35	35.00					5	2.00	20	20.00						
	¥ EHC	0.75													20	15.00				
NO	Ş ¥	0.50	96	45.00	35	17.50	15	7.50			95	47.50								
88/	S S	0.25			30	7.50	85	21,25					3	7.50	80	20.00	55	13.75		
73	VLC	0.15							20	7.50			20	7.50						
	ع ت	CARBON	55.00	00	09	60.00	38	28.75		7.50	52.50	50	35.00	00	35.	35.00	13.	13.75		
	8	0.100							50	5.00									100	10.00
	DBrL	0.075																		
8	AL.	0.050															45	2.25		
inc	LAL LAL	0.025																		
)A	MA	0.010																		
<u> </u>	RL	0.001																		
لــــــــــــــــــــــــــــــــــــــ	2=	LACQUER RATING							2	5.00							2.	2.25		
لـــا	CLEAN	0																		
	ZONA	ZONAL RATING																		
二	LOCATIC	LOCATION FACTOR																		

10.00

16.00

35.00

35.00

52.50

12.50

28.75

60.00

55.00

WEIGHTED RATING

CRC DIESEL RATING SYSTEM

,这是这个人,我们就是这种的人,我们就是这个人,我们就是这个人,我们就是这个人,我们就是这个人,我们也是一个人,我们也是一个人,我们也是一个人,我们也是一个人, 第一个人,我们就是这个人,我们就是这个人,我们就是一个人,我们就是一个人,我们就是一个人,我们就是一个人,我们就是一个人,我们就是一个人,我们就是一个人,我们就

STANDARD COMPUTATION SHEET FOR PISTON RATING

TEST LABORATORY AFLRI LUBRICANT AL-8822-L TEST PROCEDURE_ TEST HOURS___210

LABORATORY TEST NUMBER 703-22 STAND NO. 2 ENGINE NO. 3D-131703 FUEL AL-8818-F

NO. 1 GROOVE, VOLUME-%

PISTON NO.

l																PISTO	N WTD.	PISTON WTD. RATING	g	255
						GROOVES	VES							4	LANDS				S	ER.
0	DEPOSIT TYPE	FACTOR	NO.	1.1	S	NO. 2	ž	NO. 3	NO. 4	4	NO. 1	-	S	NO. 2	S	NO. 3	S	NO. 4	CROWN	Z
			AREA-%	AREA-% DEMERIT AREA-% DEM	AREA.%		AREA-%	RITAREA% DEMERIT AREA% DEMERIT AREA% DEMERIT AREA% DEMERIT AREA% DEMERIT AREA% DEMERIT AREA% DEMERIT	AREA-%	DEMERIT	AREA-%	DEMERIT	AREA-%	DEMERIT	AREA.X	DEMERIT	AREA.%	DEMERIT	AREA-%	DEMERIT
L	웃	8.			20	20.00					30	30.00	50	50.00						
	MHC	0.75																		
NO	¥	0.50	30	15.00	70	20.00	10	5.00												
887	ပ	0.25	20	17.50	07	10.00	09	15.00	5	1.25	70	17.50	52	12.50	85	21.25				
<u> </u>	VLC	0.15																		
	26 ت	CARBON RATING	32	32.50	50.	50.00	20	20.00	1.25	?5	47.50	.50	62.	62.50	21	21.25				
<u> </u>	BL	0.100					30	3.00									20	2.00	100	10.00
	DBrL	0.075																		
8	AL.	0.050																		
סחו	ם ראר	0.025							95	2.375					15	.375	80	2.00		
)A	VIAL.	0.010																		
<u> </u>	R	0.001																		
	2=	LACQUER					3.	3.00	2.375	375					.375	75	4.00	0		
	CLEAN	0																		
	ZONAL	ZONAL RATING																		
ニ	OCATIO	LOCATION FACTOR																		
نگا	FIGHTE	WEIGHTED RATING -	32	32,50	50.	50.00	23.00	00	3.6	3.625	47.50	50	62	62.50	21.625	525	4.00	_	10.00	

CRC DIESEL RATING SYSTEM

STANDARD COMPUTATION SHEET FOR PISTON RATING

PISTON NO. 1

LABORATORY TEST NUMBER 703-22 STAND NO. 2 ENGINE NO. 3D-131703 FUEL AL-8818-F

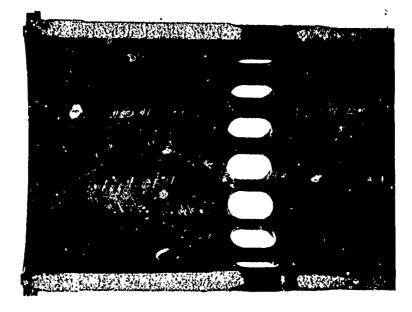
TEST LABORATORY AFLRI LUBRICANT AL-8922-L

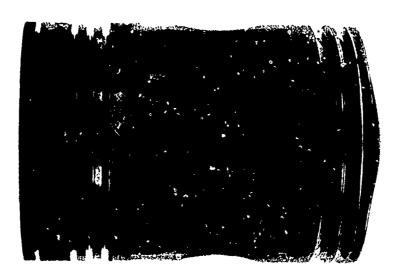
TEST PROCEDURE_ TEST HOURS___210_

454 NO. 1 GROOVE, VOLUME-% PISTON WTD. RATING

L						GROC	OOVES							LA	LANDS				S	UNDER.
Ó	DEPOSIT	DEPOSIT	Š.	1.1	ž	NO. 2	ž	NO. 3	S	NO. 4	NO. 1	_	N	NO. 2	NO. 3	.3	S	NO. 4	CRC	CROWN
	9	יאכו	AREA.%	AREA-% DEMERIT AREA-% DEME	AREA-%	DEMERIT	AREA-%	DEMERIT	AREA.X	DEMERIT	AREA-%	DEMERIT	AREA.%	DEMERIT	AREA:X	DEMERIT	AREA-%	RITAREA% DEMERIT AREA% DEMERITAREA% DEMERITAREA% DEMERITAREA% DEMERIT AREA% DEMERIT AREA% DEMERIT	AREA.%	OEMERIT
	S H	1.00	40	40.00	100	100.00					40	40.00	45	45	10	10.00				
	AHC	0.75																		
NO	₩	0.50	09	30.00			90	45.00							90	45.00				
88/	27	0.25					10	2.50			50	12.50	40	10.00			15	3.75		
()	VLC	0.15							100	15.00	10	1.50	15	2.25			75	11.25		
	J€	CARBON	70.	70.00	100	100.00	7	47.50	ï	15.00	54.	54.00	57.25	25	55	55.00	15.00	00		
<u> </u>	8	0.100																	100	10.00
	DBrL	0.075																		
<u>8</u>	Æ	0.060																		
no	OO!	0.025															21	.25		
AC	Y.A.	0.010																		
1	RL	0.001																		
	2=	LACQUER															. 2	.25	10.	10.00
لت	CLEAN	0																		
	ZONAL	ZONAL RATING						i												
1_	OCATIO	LOCATION FACTOR																		
_	VEIGHT!	WEIGHTED RATING	70	70.00	10(100.00	47	47.50	15	15.00	54.	54.00	57.25	25	55.00	00	15.25	25	10	10.00

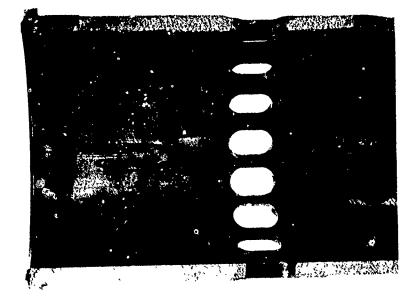
PISTON AND CYLINDER LINER CONDITION Test #22

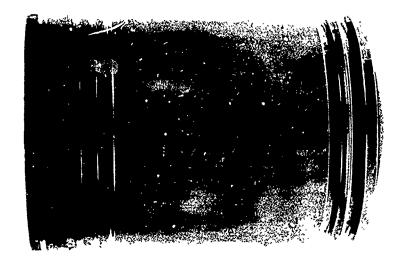




No. 2 - Thrust Side (Best)

PISTON AND CYLINDER LINER CONDITION Test #22



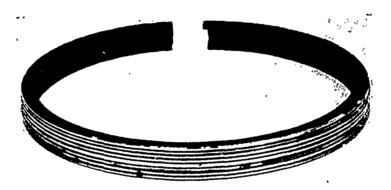


No. 1 - Anti-Thrust Side (Worst)

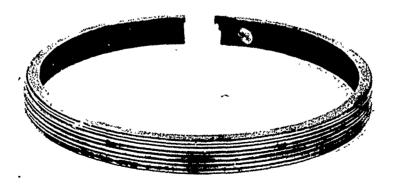
RING FACE CONDITION Test #22



Piston - 1



Piston - 2



Piston - 3

APPENDIX C

U.S. ARMY TRACKED-VEHICLE TEST PROCEDURE

6V-53T 240-HOUR TRACKED VEHICLE CYCLE ENDURANCE TEST

Test Number: Test Fuel: Test Lubricant:

Sequence of Events

- Disassemble, rebuild, measure, and record the engine rebuild measurements as provided for in the BEFORE AND AFTER TEST DATA SHEETS.
- Prior to the test, calibrate: tachometer
 flowmeter
 load measurement system
 temperature indicators (1-7)
 and exhaust temperature (1 and 2)
 pressure indicators
- Break engine according to sections 6.1 to 6.3.3 of endurance test procedure.
- Perform initial prior calibration check according to section 6.4 of the test procedure.
- Perform shake-down run according to section 6.5.
- Perform final power calibration check according to section 6.6.
- Perform before-test full load performance determination according to sections 7.1 to 7.3.
- Perform 240-hour endurance test according to sections 7.4 to 7.12.2.
- Perform after-test fuel load performance determination according to sections 7.10 and 7.11.
- Dissemble engine and remeasure the pieces provided for in the BEFORE TEST AND AFTER TEST DATA SHEETS.
- Rate the engine for deposits according to applicable CRC methods.
- Photograph engine parts specified on the analytical lab request form.

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6V-53T 240-HOUR TRACKED VEHICLE CYCLE ENDURANCE TEST PROCEDURE

PERFORMANCE OF ENGINE LUBRICATING OILS IN A TWO-CYCLE DIESEL ENGINE UNDER CYCLIC, TURBO-SUPERCHARGED CONDITIONS

1. SCOPE

- 1.1 This method is used for determining the effect of lubricating oils on ring-sticking, wear and accumulation of deposits in a reciprocating internal combustion engine. Evaluation is based on: (a) the ability of the test engine to maintain performance throughout the cycle, (b) wear developed in critical engine components, (c) accumulation of fuel and lubricant related engine deposits, particularly in the piston ring zone areas, and (d) the physical and chemical condition of the lubricant monitored throughout the test.
- 1.2 The test involves the operation of a military six-cylinder, fuel injected, turbo-supercharged, 2-stroke-cycle diesel engine under cyclic conditions for a total of 240 hours. Prior to test the engine is reconditioned as described herein. Evaluation is made by comparing the test oil performance to that of a reference oil of known quality.

2. SAMPLE

- 2.1 A minimum of 55 gallons of test oil is required.
- 5. PREPARATION FOR TEST.
- 5.1 Engine Disassembly. A systematic inspection and maintenance of the test engine shall be performed prior to each test run. New engines or engines being used for the first time in this test method and thereafter will be disassembled, reconditioned, and gauged after each test. Regardless of their condition, the following parts shall be replaced with new factory production items:

Piston Assemblies
Piston Ring Sets
Cylinder Liners
Fuel Filters, Oil Filter, and Air Filters
All Gaskets and Seals

- 5.2 Cleaning Procedure.
- 5.2.1 Engine Block. If the engine is completely disassembled, the block shall be cleaned by spraying with solvent.
- 5.2.2 Aluminum Parts. Aluminum parts will be cleaned by spraying with solvent followed by air drying. If deposits are stubborn, the parts may be soaked in solvent for a period up to two hours at a temperature of 100°F or less. The solvent soak must be followed by a warm water wash and air drying.
- 5.2.3 Steel Parts. All steel parts (i.e., rocker arm covers, oil pan, cylinder head decks, oil pump, crankshaft, etx.), shall be cleaned by spraying with solvent, air dried and lightly coated with reference oil.

- 5.2.4 Fuel Injectors. The fuel-injectors are removed but not disassembled or adjusted. Only the tips should be lightly wire brushed to remove carbon particles. Should the operation of the engine indicate that their condition might be at fault, the units should be tested, adjusted, and/or replaced with new units.
- 5.2.5 Combustion Chambers and Valves. Exhaust valves are removed and the entire combustion chamber area of each cylinder is cleaned by wire brushing. Valves are only lightly refaced, if inspection shows pitting. Where the sealing surfaces (faces) are not pitted, the valves need be only lightly lapped prior to reassembly. If light refacing does not correct the seating condition, the valve shall be replaced.
- 5.3 Engine Disassembly. The engine block is fitted with new parts as listed in sections 3.2.1 and 5.1. Complete measurements of the block bore, liners, pistons, rings, connecting rod journals, main bearing journals, connecting rod bearing inserts, and main bearing inserts are made prior to each rebuild. Connecting rods and piston pins will be inspected and replaced, if not in good service condition. In addition, camshaft journals to bearings and oil pump clearances shall be checked against service limits prior to every test. These parts will be replaced as required to maintain service limits. The liner outside surfaces contacting the block bore shall be lightly coated with grease to reduce interface fretting corrosion. Other parts shall be coated with reference oil during assembly.

5.3.1 The following critical rebuild measurements shall be maintained during engine assembly:

Engine Part	Tolerance or Clearance (1)-Inches
Crankshaft main bearing clearance	0.0011-0.0041
Camshaft bearing clearance	0.0045-0.0060
Connecting rod bearing clearance	0.0010-0.0040
Crankshaft end-play	0.0040-0.0120
Cylinder block bore	
Taper	0.0015 max
Out-of-round	0.0015 max
Inside diameter	4.3565-4.3575
Clearance liner to block	0.0000-0.0025
Cylinder liners (installed)	(2)
Taper	0.0015 max(2)
Out-of-round	0.0015 max (2)
Inside diamter	3.8753-3.8767
Piston to liner fit	0.0060-0.0095
Piston skirt 0.D.	3.8752-3.8767
Fire Ring	
End gap	0.020-0.046
Side clearance	0.003-0.006
#1 Compression ring	
End gap	0.020-0.045
Side Clearance	0.007-0.010
#2 Compression ring	
End gap	0.020-0.046
Side clearance	0.005-0.010
#3 Compression ring	
End gap	0.020-0.045
Side clearance	0.005-0.010
0il rings	
End gap	0.010-0.025
Side clearance	0.0015-0.0055

1/All tolerances and clearances given in inches. 2/Using new cylinder liners in a used block.

- 5.3.2 Engine assembly shall be in accordance with TM 9-2815-212-35. and the Detroit Diesel Engine Series 53 Service Manual. Reference must be made to this document to determine the proper bolt torques, tightening sequences, and final injector timing and valve lash settings.
- 6. ENGINE CONDITIONING AND CALIBRATION PROCEDURES
- 6.1 Engine start-up and shut-down procedures.
- 6.1.1 Engine Start-Up Procedure. From a cold start, idle the engine for five minutes. Then warm up at 1200 rpm and 20 lb dynamometer load* (20 BHp) until oil sump temperature reaches 180°F and coolant jacket-out temperatures reaches 175°F. If the coolant system was drained at the previous shut-down, warm up at 1100 rpm and 16 lb load (15 BHp) to insure deaeration of the coolant system. If the engine is started warm and the 180°F oil sump and 175°F coolant

jacket-out temperatures are achieved, it is permissible to gradually accelerate the engine without delay to test conditions. The automatic controller set point for coolant out temperature must remain at 170°F during all startups except when the test is being resumed at the one hour idle modes described in 7.4.

- 6.1.2 Engine Shut-Down Procedure. To shut down the engine from test conditions, slowly bring the engine to idle by turning the rack setting to the idle position. Allow the engine to idle for five minutes and then shut-down by actuating the idle cut-off. The automatic controller set point for coolant out temperatures must remain at 170°F during all shut-downs.
- 6.2 Engine Run-In Procedure.
- 6.2.1. Oil Charge. Charge the engine oil sump with approximately twenty 28 quarts of tes oil. Disconnect the turbo inlet oil supply line and crank the engine with the governor control in the fuel cut-off position until one pint of oil is pumped from the disconnected line. Reconnect the turbo line and crank the engine until the oil pressure stabilizes.
- 6.2.2. Operating Conditions. Start the engine in accordance with 6.1.1. and conduct the engine run-in according to the following schedule:

	Dynamometer		
Engine Speed, rpm	Load, 1b	Power, Obs BHp	Time, min.
1800	19	30	15
2200	69	130	30
2500	93	200	30
2800	94	225	30

Coolant jacket-out temperature is maintained at $170 \pm 2^{\circ}F$, and oil gallery pressure is 40 psi minimum. Coolant system deaeration (air free sight glass) should be established by the time the 2500 rpm sequence is completed.

- 6.3 Interim Settings and Adjustments.
- 6.3.1. Immediately following the run-in; check, adjust and record the governor settings. Set the idle speed at 650-700 rpm maintaining a minimum of 5 psi gallery oil pressure. No-load speed should be 2950-3030 rpm.
- 6.3.2. Shut down the engine according to 6.1.2.
- 6.3.3. Five minutes after shut-down, check and reset the injector timing and exhaust valve clearance as follows:

Injector timing - 1.4600 ± 0.0035 inch Valve clearance - 0.023 to 0.25 inch (hot)

- 6.4 Initial Power Calibration Check. Full-rack power calibration checks are made in order at 2200, 2500 and 2800 rpm. Engine start-up is in accordance to
- 6.1.1. The engine is operated at the specified speed until the observed output has stabilized. A coolant jacket-out temperature os $170^{\circ} \pm 2^{\circ}F$ is

^{*} Loads specified are for a dynamometer with a dynamometer constant of 1167.

maintained through calibration checks. The parameters shown below with be within specified limits.

Calibration Parameter	2200	2500	2800
Minimum Observed Output, 1b (Bhp) Normal Oil Gallery Pressure, psi Minimum Oil Gallery Pressure, psi	30	124(265) - 40-60 - 32 60 -	32
Maximum Oil Gallery Pressure, psi Minimum Air Box Pressure, in. Hg Maximum Air Inlet Restriction, in. H ₂ O Maximum Crankcase Pressure, in. H ₂ O Maximum Exhaust Back Pressure, in. Hg	4-5 4 3 2•1	5-8 5 4.5 2.7	
Normal Fuel Pressure, psi Minimum Fuel Pressure, psi		- • •	

- (1) Deduct 4 1b dynamometer load (2 BHp) for each 10°F rise in ambient intake air temperature above 85°F (moist air).
- 6.5 Shake-Down Run. Immediately following the 2200 rpm power calibration check, the engine is operated for a period of five hours at 2800 rpm and 104 lb. dynamometer load (250 BHp) maintaining a coolant jacket-out temperature of $170 \pm 2^{\circ}\text{F}$. The test parameter data listed in 7.12.1 shall be recorded hourly during the shake-down run.
- 6.6 Final Power Calibration Check. Following the shake-down run, full-rack power calibration checks shall be made in accordance with 6.4. On completing the 2200 rpm power check, the engine will be shut-down in accordance to 6.1.2 and an airbox inspection performed (see 7.9).

7. TEST PROCEDURE

- 7.1 Oil Drain. After final power calibration shut-down and while the oil is still warm, drain the oil and remove the oil filter.
- 7.2 Oil Charge. Weigh-in a new oil filter and sufficient test oil to bring the sump level to the full mark on the dipstick gage (approximately twenty-eight quarts). Recheck oil level after cranking engine long enough to stabilize the oil pressure.
- 7.3 Full Load Performance Determination. Conduct a full load (full rack) performance determination measuring engine dynamometer load, brake horsepower (BHp), and brake specific fuel consumption (BSFC) at 200 rpm intervals between engine speeds of 1800 and 2800 rpm. Record all data as per 7.12.1.
- 7.4. Test Duration. The test consists of 240 hours operation at prescribed test conditions. Interim oil adjustments, airbox inspections and oil samplings are made on the following schedule:

					Hot	ırs (of O	pera	tion				
Operation	0	20	40	60	80	100	120	140	160	180	200	220	240
Oil Adjustments		X	X	X	X	X	X	X	X	X	X	х	_
Used Oil Sampled	-	X	X	*	X	X	*	X	X	*	X	X	*
Airbox Inspected	X	-	-	X	-	-	X	-	~	X		-	-
Oil Change	X	-	~		-	-	X	-	-		-		-

X indicates adjustment, sampling or inspection to be performed at given test time.

7.5 Test Cycle Description. The endurance test consists of repeating a four-mode, five-hour operating cycle four times daily for a total of 20 hours. The engine is then shut down for a period of four hours after which the daily cycle is repeated. The five-hour operating cycle (shown below) consists of: 0.5 hour at engine idle followed by 2.0 hours at maximum power, followed by 0.5 hour at engine idle followed by 2.0 hours at maximum torque. The 20-hour endurance cycle is conducted for 12 days without interruption.

Endurance Test Operating Cycle Jacket-Out °F Speed, rpm Period Mode Time, hrs Temp, Idle 650 ± 25 2800 ± 20 0.5 100 2 170 Max Power 100 Idle 0.5 650 100 0 Max Torque 2 100 2200 + 20170 2 0.5 0 650 100 Idle Max Power 100 2800 2 170 Idle 0.5 650 100 0 Max Torque 2 100 2200 170 3 100 Idle 0.5 0 650 100 Max Power 2 2800 170 Idle 0.5 0 650 100 Max Torque 2 100 2200 170 100 Idle 0.5 0 650 Max Power 2 100 2800 170 Idle 0.5 650 100 0 100 2200 170 Max Torque 2 5 0 Shutdown 0

7.6 Performance Equations:

Horsepower = $(Load \times Speed)/1167$

BSFC = (Fuel Consumption)/Horsepower

^{*} indicates 32 oz.(1 qt) samples.

- 7.7 <u>Used Oil Sampling.</u> Take a 8 oz. sample of oil at the oil filter housing according to the schedule specified in 7.4. The 60, 120, 180, and 240-hour samples are each 32 oz. samples. This is done with the engine idling prior to the scheduled shut-down and oil adjustment.
 - 7.8 Oil Adjustment and Oil Change.

- 7.8.1 Shut down the engine according to 6.1.2. Note that oil samples must be taken during the 5 minute idling period (see 7.7).
- 7.8.2 Make oil adjustments according to the schedule of 7.4 by adding a weighed amount of test oil to the sump, bring the level to the full mark on the dipstick. Maintenance of an oil log sheet is required. If the oil level is below add halfway (10 hours) through the period, weight and add roughly a gallon of oil so that the engine can finish the period safely.
- 7.8.4 After the 120-hour oil sample is taken, the engine is shut down according to 6.1.2. The oil is drained from the oil filter housing and the engine crankcase. A new charge of test oil and a new oil filter are installed per 7.2.
- 7.9 Airbox inspections. Four airbox inspections shall be made as specified in 7.4. The zero-hour inspection is made at the completion of the final power check and prior to the installation of test oil. Observations made at each airbox inspection must be recorded and included in the final engine inspection. The areas inspected, performance levels noted and means of inspection shall be as follows:

١,

Area Inspected Inlet Ports	Performance Level Noted Percent Plugging	Means of Inspection Visual
Piston Skirt	Tinplate melting Scoring	Visual
	Burning	
Ring Lands	Carbon Deposits	Visual
Rings	Freedom	Blunt Probe
	Face Scuffing	and
	Face Burning	Visual
Cylinder Liner	Scuffing 1/	Illuminated
	Scoring	and Magnifying
	Bridge Cracking	Borescope

1/Scuffing shall be described in terms of degree (light, medium and heavy) and in terms of area (thrust, anti-thrust, front and rear).

- 7.10 Full Load Performance Determination. Immediately following the final endurance cycle, conduct a full load (full rack) performance determination in the same manner as described in 7.3 (Before Test).
- 7.11 Final Oil Drain. Shut down engine as outlined in 6.1.2. Let the engine stand for five minutes, then drain the crankcase and oil filter housing. Weigh and record the quantity of oil drained and oil filter.

APPENDIX D

240-HOUR TRACKED-VEHICLE CYCLE USING 6V-53T DIESEL ENGINE

Test Lubricant: AL-9065-L (MIL-L-21260B)
Test Fuel: 1% S DF-2 (AL-9242-F)
Engine Test Number: HSF-1
Pate Completed: 7 July 1980

Conducted For

U.S. Army Mobility Equipment Research and Development Command Energy and Water Resources Laboratory Ft. Belvoir, Virginia

by

U.S. Army Fuels and Lubricants Research Laboratory Southwest Research Institute San Antonio, Texas 78284

6V-53T

TEST: HSF-1

ENGINE REBUILD MEASUREMENTS
Model Number: 5063-5397
Serial Number: 6D-178671

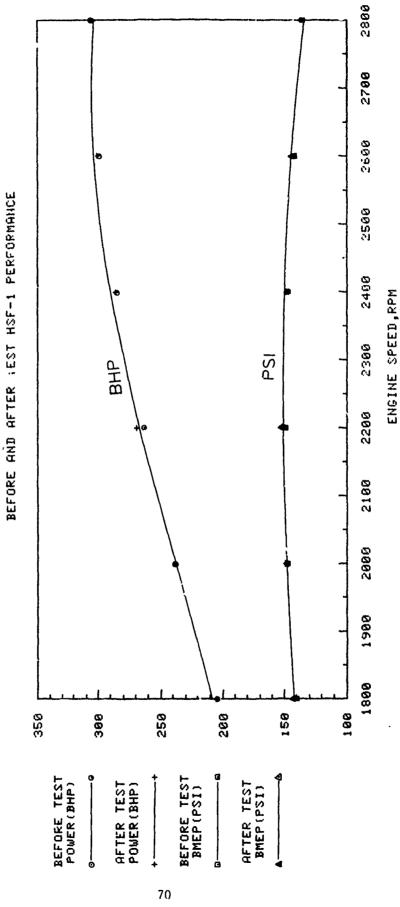
	Min	Max	Avg	Specified Limits*
Cylinder Block Bore	/ 25/0	. 2570	4.3573	4.3595 max
Inside Diameter (bottom)	4.3568 0.0001	4.3578 0.0008	0.0005	0.0015 max
Out-of-Round	0.0001	0.0006	0.0003	0.0015 max
Taper	0.0000	0.0000	0.0003	0.0015 max
Cylinder Liners (installed)				
Inside Diameter	3.8755	3.8769	3.8760	3.8752-3.8767
Out-of-Round	0.0000	0.0010	0.0003	0.0020 max
Taper	0.0000	0.0013	0.0005	0.0010 max
•				
Piston Diameter				
(@ skirt)	3.8680	3.8682	3.8681	3.8669-3.8691
Piston Skirt to Cylinder				
Liner Clearance	0.0074	0.0089	0.0080	0.0061-0.0098
Compression kings	0.005	0.020	0.037	0.020-0.046
Gap (Fire Ring)	0.035	0.038	0.037	0.020-0.046
Gap (Others)	0.024	0.037	0.030	0.020-0.030
Ring-to-Groove Clearance	0.000	0.004	0.004	0.003-0.006
Top (Fire)	0.003	0.004	0.009	0.003-0.000
No. 1	0.008	0.009	0.009	0.005-0.008
No. 2 and No. 3	0.006	0.007	0.007	0.005-0.000
Oil Control Pinco				
Oil Control Rings	0.019	0.024	0.021	0.010-0.025
Gap	0.002	0.003	0.003	0.0015-0.0055
Ring-to-Groove Clearance	0.002	0.003	0.003	0.0015 0.0010
Connecting Rod Bearing- to-Journal Clearance	0.0022	0.0025	0.0024	0.0011-0.0041
to-Journal Clearance	0.0022	0.0025		
Main Bearing-to-Journal Clearance	Not removed	or rebuilt	for this test	0.0011-0.0041
Camshaft Bearing-to-				0.0045.0.0060
Shaft Clearance	Not removed	or rebuilt	for this test	0.0045-0.0060

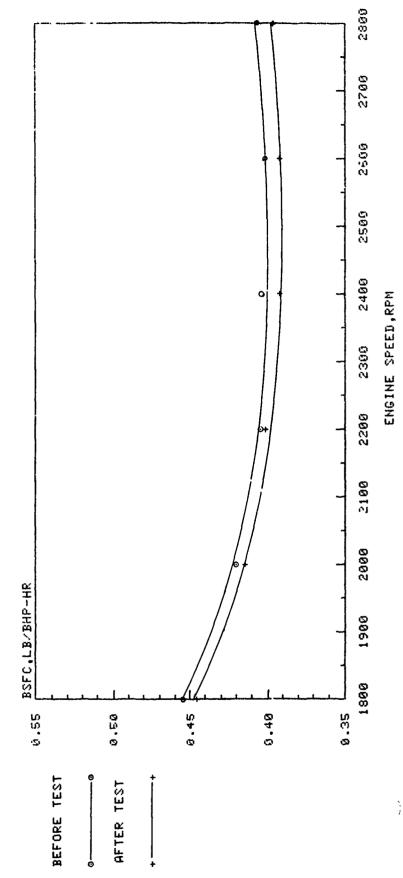
^{*}All measurements are in inches.

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6V-53T 249 HOUR TRACKED VEHICLE CYCLE BEFORE AND AFTER LEST HSF-1 PERFORMANCE

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6V-53T 240-HOUR TRACKED VEHICLE CYCLE ENDURANCE TEST TEST HS"-1 OPERATING CONDITIONS SUMMARY

Lubricant: AL-9065-L Fuel: 1% S DF-2

	Maximum Po	•	Maximum Torque Mode (2200 rpm)		
	Mean Standard* Deviation		Mean	Standard Deviation	
Engine Speed, rpm Torque, Ft-1b (N-m) Fuel Consumption, 1b/hr (kg/hr) Observed Power, Bhp (kW) BSFC, 1b/Bhp-hr (g/kW-hr)	2802 585(793) 124.4(56.48) 309(231) 0.401(244)	2 2(3) 1.0(0.45) 1(1) 0.003(2)	2203 643(872) 107.1(48.62) 269(201) 0.397(241)	4 4(5) 0.6(0.27) 2(1) 0.003(2)	
Temperatures, °F (°C) Exhaust before turbo Exhaust after turbo Water Jacket Inlet Water Jacket Outlet Oil Sump Fuel at Filter Inlet Air (at compressor) Airbox	1008(542) 820(438) 156(69) 171(77) 244(118) 94(34) 95(35) 302(150)	14(8) 9(5) 1(1) 1(1) 1(1) 2(1) 3(2) 4(2)	1041(561) 856(458) 155(68) 170(77) 236(113) 91(33) 94(34) 248(120)	13(7) 13(7) 1(1) 1(1) 2(1) 2(1) 2(1) 4(2)	
Pressures Exhaust before turbo, psi (kPa) Exhaust after turbo, in. Hg (kPa) Compressor Discharge, psi (kPa) Blower Discharge, psi (kPa) Oil Gallery, psi (kPa) Intake Vacuum, in. H ₂ O (kPa)	15.3(105) 2.8(9.5) 14.6(101) 20.4(141) 42(289) 8.1(2.0)	0.3(2) 0.1(0.3) 0.3(2) 0.4 1(7) 0.5(0.1)	10.3(71) 1.8(6.1) 11.0(76) 12.4(85) 38(262) 4.9(1.2)	0.2(1) 0.1(0.3) 0.3(2) 0.3(2) 1(7) 0.3(0.1)	
Ambient Conditions (both modes of operation) Dry Bulb Temperature, °F (°C) Wet Bulb Temperature, °F (°C) Barometric Pressure, in. Hg (kPa)	87(31) 83(28) 29.08(98.58)	6(3) 5(3) 0.15(0.51)			

^{*}68% of the values for a given variable occur within \pm 1 standard deviation of the mean; 95% occur within \pm 2 standard deviations.

6V-53T TEST: HSF-1 Lubricant: AL-9065-L

LUBRICANT ANALYSIS

	ASTM		Test Time, Hours			
Test <u>Method</u>	Test Method	New Oil	_60	120	180	240_
Kinematic Viscosity @ 40°C,						
cSt	D 445	106.8	115.6	118.2	114.9	117.3
Kinematic Viscosity						
a 100°C, cSt	D 445	11.7	12.5	12.7	12.4	12.6
Viscosity Index	D 2270	97	99	99	98	99
Total Acid Number, mg KOH/g	D 664	2.2	2.4	2.7	2.4	2.6
Total Base Number, mg KOH/g	D 664	9.4 *	4.43	4.73	4.13	4.43
Pentane B Insolubles, wt%	D 893	ND	ND**	0.19	ND	0.20
Toluene B Insolubles, wt%	D 893	ND	ND	0.17	ND	0.18
Flash Point, °C (°F)	D 92	246(475)	244(471)	246 (475)	246(475)	244(471)
Density, gm/ml @ 16°C						
(60°F)	D 287	0.888	ND	0.889	ND	0.886
Carbon Residue, wt%	D 524	1.28	ND	1.67	ND	1.67
Sulfated Ash, wt%	D 874	1.41	ND	1.49	ИD	1.58
Other		a				

^{*}TBN by D 2896 = 10.6 .

**Properties not deteremined at these test times.

a = new contained 0.15%w Zn

^{0.36%}w Ca

^{0.23%}w S 0.08%w P

⁶² ppm N

6V-53T TEST: HSF-1 Lubricant: AL-9065-L

TOTAL OIL CONSUMPTION AND WEAR METALS BY XRF

		_Wear Me	tals+,ppm
Test Time, Hours	Total Oil Consumed, 1b (kg)	Iron	Copper
20	13 (5.9)	76	27
30.5	20 (9.1)	_++	_
40	27 (12.2)	97	11
53	34 (15.4)	_	-
60	40 (18.1)	83	16
65.5	45 (20.4)	-	-
80	57 (25.9)	84	24
90.5	65 (29.5)		- .
100	71 (32.2)	81	ND*
110	78 (35.4)	-	-
120	88 (39.9)	84	ND
140	94 (42.6)	45	ND
150	101 (45.8)	-	-
160	108 (49.0)	52	ND
170	115 (52.2)	-	-
180	118 (53.5)	49	ND
190	125 (56.7)	-	-
200	132 (59.9)	61	ND
210	139 (63.0)	-	-
220	145 (65.8)	66	ND
228	152 (68.9)	-	••
240	161 (73.0)	66	ND

*Not detected.

^{*}No other wear metals detected. **Hoil samples for wear metal analysis not taken at these times.

6V-53T

TEST: HSF-1

Lubricant: AL-9065-L

WEAR MEASUREMENTS

Cylinder Liner Bore Diameter Change*

Cylinder Number

		11.	2	L	_3L	
	<u>T-AT**</u>	F-B	T-AT	F-B	T-AT	F-B
Тор	0.0008	-0.0001	0.0010	0.0006	0.0008	0.0005
Middle	0.0008	0.0005	0.0009	0.0003	0.0007	0.0008
Bottom	0.0005	0.0005	0.0007	0.0007	0.0014	0.0005

Cylinder Number

	_	1R_	_ 2	R_	_3R	_
	T-AT	F-B	<u>T-AT</u>	F-B	T-AT	<u>F-B</u>
Тор	0.0008	0.0004	0.0011	0.0007	0.0014	0.0001
Middle	0.0007	0.0012	0.0012	0.0004	0.0009	0.0007
Bottom	0.0005	0.0005	0.0005	0.0010	0.0005	0.0002

Average Change

	<u>T-AT</u>	<u>F-B</u>
Тор	0.0010	0.0004
Middle	0.0009	0.0007
Bottom	0.0007	0.0006

T-AT Average Change: 0.0009 Overall Average Change: 0.0007

Piston Ring End Gap Change

Ring Number	<u>1L</u>	<u>2L</u>	_3L_	<u>1R</u>	<u>2R</u>	_3R_	Average Change
1(F/R)	0.008	0.014	0.014	0.006	0.008	0.009	0.010
2	0.000	0.000	0.001	0.001	0.003	0.001	0.001
3	0.000	0.000	0.000	0.001	0.002	0.001	0.001
4	0.000	0.001	0.002	0.003	0.003	0.003	0.002
5	0.006	0.009	0.009	0.009	0.011	0.009	0.009
6	0.005	0.008	0.008	0.008	0.009	0.009	0.008
7	0.005	0.006	0.006	0.006	0.008	0.008	0.007

Overall Average Change: 0.005 Average F/R Δ = 0.010 IN

^{*}All dimensions given are in inches.
**T-AT = Thrust-Antithrust Direction; F-B = Front-Back Direction.

6V-53T TEST: HSF-1 Lubricant: AL-9065-L

POST TEST ENGINE CONDITION AND DEPOSITS

Cylinder Liner

 . ,			Cylin	der Numb	er		
	11.	2 <u>L</u>	3L	1R	2R	<u>3R</u>	Average
Intake Port Plugging, % restriction	5	1	1	1	1	1	2
Liner Scuffing, % Area [†] Thrust Anti-Thrust Total	10 5 7.5	15 20 17.5	20 5 12.5	15 5 10	10 5 7.5	5 10 7.5 Overall:	13 8
Liner Glazed, % Area	20	15	10	15	25	15	17
Liner Lacquer, % Area	40	40	50	40	30	40	40
B. Pistons							
Ring Face Burn, % Area Fire Ring No. 1 No. 2 No. 3	10 10 5 5	20 1 2 2	5 35 55 75	5 2 5 2	3 5 10 2	70 25 0 15	19 13 13 17
						Overall:	16
Ring Groove Carbon, % Fi Fire Ring No. 1 No. 2 No. 3	11ing 20 75 5 0	20 80 5 0	65 85 25 0	60 90 5 0	20 90 20 0	15 90 10 0	33 85 12 0
						Overall:	33
Piston Skirt Deposit Rat (Demerit)*	ing						
Thrust Anti-Thrust	5.4 LS 5.8 S	6.1 LS 6.1 LS,PM	5.5 LS 6.0 LS	5.8 I.S 6.0 LS	6.2 LS 6.0 LS	5.8 LS 5.8 LS	5.8 LS 6.0 LS
						Overall:	5.9

^{*}LS = Light scratch s, S = scratches, PM = plate melt.
+Light vertical lines present in all cylinders not counted as scuffing.

6V-53T TEST: HSF-1 Lubricant: AL-9065-L

POST TEST ENGINE CONDITION AND DEPOSITS

B. Pistons (continued)

			Cyline	der Numb	<u>er</u>		
	<u>IL</u>	<u>2L</u>	3L	<u>1R</u>	2R	<u>3R</u>	Average
Oil Control Ring Grooves (Demerit)							
Upper	3.0	4.0	3.0	4.0	4.0	4.0	3.7
Lower	3.0	4.0	3.0	4.0	4.0	4.0	3.7
						Overall:	3.7
Piston Groove Inside Diameter, % Ring Supporting Carbon							
No. 1	0	0	0	4	0	0	1
No. 2	44	78	55	75	85	89	71
						Overall:	36
Piston WTD Rating	363	375	400	420	403	353	386
Ring Sticking No. 1 No. 2 No. 3 No. 4 Piston Oil Drain Holes			A11 A11	l Rings Rings Rings Rings Rings	Free Free		
C. Exhaust Valves							
Deposits Head Face Tulip Stem			- medium BHO	C to soot	to No. 9 t	lacquer	
Stem Condition Freeness in Guide Head Face Seat Stem Tip	1.1.*	L	L No	Free ormal ormal nal Wear ormal	I.	L	

^{*}LL = light leaking, L = leaking.

6V-53T TEST: HSF-1 Lubricant: AL-9065-L

POST TEST ENGINE CONDITION AND DEPOSITS

D. Other Ratings

			<u>Cy1</u>	linder Numbe	<u>r</u>		
	<u>ll</u>	_2L_	_3L	<u>1R</u>	<u>2k</u>	_3R_	
Tappets, Cams, and Rocker Arms				61			
Tappet Deposit Tappet Surface				Clean			
Condition Cam Lobes	RSCR ⁺	N	RSCR	Normal	N	N	•
Rocker Arms				Normal			_
Tip Bushing				Normal			-
Shaft				Normal			•
Bearing Surface Condition							
Main Bearings Main Journals				rance tests) Normal			
Rod Bearings	Replaced	(after 3	endur	cance tests)	due to	fatigue	
Rod Journals Piston Pin	LG	LG	MG	Normal	MW	LG	•
Pin Bushing				out of spe			•
Combustion Chamber							
Deposits					10%	108	
G F			15%	10%	10% 5%	10%	
E	10%	15%	5%	10%	570		
D				5%			
С		10%	5%	5%	35%	15%	
В	20%	10%	20%	5%	10%	10%	
A	10%	50%	40%	40%	25%	25%	
¹ ₂ A							
¹cA	60%	15%	20%	25%	15%	40%	

⁺RSCR = Rear lobe scratched, N = normal.

STANDARD COMPUTATION SHEET FOR PISTON RATING

TEST HOURS 240
TEST LABORATORY AFLRE TEST PROCEDURE 240 LUBRICANT_AL=9065=L

LABORATORY TEST NUMBER H.S.F. -1
STAND NO. 5 ENGINE NO. 6D-178671
FUEL AL-9242-F 1750F?

PISTON NO. -

NO. 1 GROOVE, VOLUME-%

5	SRICA	LUBRICANT_AI=9065=I	1065-1			5		777			ļ								+	
ı I	ı	i														PISTON	WTD.	PISTON WTD. RATING	-	363.
1						GRO	OOVES							LA	LANDS				UNDER.	ER.
90	DEPOSIT	DEPOSIT	Ş		Ž	NO 2	_	CN CN		NO. 4	S S		NO. 2	2	NO. 3	n	NO.	*	CROWN	ž
		FACTOR	A DE A K INCLUEDIT A REA & INFINER	TIGALI	A B F A.%	DENE	AIT AREA	* DEMER	TAREA	TAREA-% DEMERIT AREA-% DEMERIT	AREA-%	AREA-X DEMERIT AREA-X DEMERIT AREA-X DEMERIT AREA-X DEMERIT AREA-X DEMERIT	AREA-%	EMERIT	AREA-%	EMERIT	AREA.%D	EMERITA	REA.%	EMERIT
	9	8	Ancw. An		6						100	100,	8	90.06						
	اد	-	1) N	30.00	 	-	-											
	MHC	0.75					+	_	+	_							+	+	1	
NO	Ç ¥	0.50	40	20.00			-	_	_	$\frac{1}{1}$							+		+	
887	ΓC	0.25		15,00	20	5.00	a		_	_			9	2.50			+			
∀ O	VLC	0.15					100	15.00	g	_					100	15.00	1			
	∞ت	CAREON	35.00	00	∞	85.00		15.00			10(100.00	92	92.50	15	15.00				
	81	0.100					_		-											
	DBrL	0.075							2	3.75							75	5.625	200	7.50
R	A P	0.050							20	2.50				_			25	1.250		
JUE	ZOE	.0.025						_	_		_						1		+	
DA	<u>۲</u>	0.010				_	_	_	_	_							+			
1_	R	0.001				_	-	_	-		_								7	
	3.	LACQUER RATING								6.25							6.875	7.5	7.50	g
	CLEAN	0				_	+	_	\dashv		-									
	ZONA	ZONAL RATING																		
	OCATI	LOCATION FACTOR							\dashv		-									
	WEIGHT	WEIGHTED RATING		35.00		85.00	\dashv	15.00	\dashv	6.25	97	100,00	92	92.50	15.00	00	9	6.875	7.50	9
١																				

The state of the s

STANDARD COMPUTATION SHEET FOR PISTON RATING

DATE 9 July 1980 H.S.F. -1 STAND NO. 5 ENGINE NO. 6D-178671 LABORATORY TEST NUMBER__ FUEL_AL-9242-F 1%SDF2 RATER LYONS TEST LABORATORY AFLRE LUBRICANT AL-9065-L **TEST PROCEDURE**. TEST HOURS_

PISTON NO. L-2

NO. 1 GROOVE, VOLUME-%
PISTON WTD* RATING 375

AREAN DEMENITAREAN 7.50 UNDER. CROWN 7.50 7.50 100 .50 4.75 **N**0. 5.25 5.25 95 17.50 2.25 19.75 19.75 NO. 13 2 LANDS 30 7.50 70 70.00 77.50 NO. 2 77.50 30 100 100.00 100.00 100,00 NO. 1 2.00 NO. 4 5.00 5.00 100 15.00 15.00 15.00 NO. 3 100 GROOVES 100 100.00 100.00 NO. 2 100.00 40.00 5.00 45.00 NO. 45.00 80 20 LOCATION FACTOR WEIGHTED RATING DEPOSIT DEPOSIT
TYPE FACTOR 0.075 0.010 0.100 0.050 0.025 0.001 ZONAL RATING 0.50 . 8 0.75 0.25 VLC 0.15 LACQUER CARBON 0 F F F DBrL CLEAN ပ္ MHC Š 뮕 81 A CARBON

WEIGHTED TOTAL DEPOSITS

STANDARD COMPUTATION SHEET FOR PISTON RATING

PISTON NO. 1-3

TEST PROCEDURE 240
TEST HOURS 240
TEST LABORATORY AFLRL
LUBRICANT AL-9065-L

RATER 1.1.40NS DATE 9 July 1980
LABORATORY TEST NUMBER H.S.F.-1
STAND NO. 5 ENGINE NO. 6D-1.78671
FUEL AI-9242-P 1.250P2

NO. 1 GROOVE, VOLUME-%
PISTON WTD* RATING 400

															-					
L						GROOVES	VES							LAF	LANDS				200	UNDER.
Õ	DEPOSIT	DEPOSIT	NO. 1		N	NO. 2	Ž	NO. 3	2	NO. 4	NO.	-	NO. 2	.2	NO. 3	.3	NO.	. 4	CRC	Z.
	7 7	TAC 108	AREA-%	AREA-% DEMERIT AREA-% DEME	AREA-X	DEMERIT	AREA.%	DEMERIT	AREA.X	DEMERIT	AREA-%	DEMERIT	AREA-%	ERITAREAM DEMERIT AREAM DEMERIT AREAM DEMERIT AREAM DEMERIT AREAM DEMERIT AREAM DEMERIT AREAM DEMERIT	IREA.K.	EMERIT	AREA-%	DEMERIT.	AREA.%	DEMERIT
	¥	1.8			100	100.00					100	100.00	85	85.00						
	MHC	0.75					10	7.50												
NO	Ş	0.50	75	37.50			25	12.50												
8A/	ပ္	0.25	25	6.25			65	16.25					15	3.75	100	2.50				
CV 8:	VLC V	0.15															72	10.50		
	3≩	CARBON	43	43.75	100.00	00	36.25	25			100.00	0	88.	88.75	2,	2.50	10.50	50		
<u></u>	BP.	0.100															30	3.00		
	DBrL	0.075							100	7.50									100	7.50
8	AL.	0.050																		
JUC	JA Z	0.025																		
)JA	3	0.010																		
	R.	0.001																		
	5*	LACQUER								7.50							3.00	8	'	7.50
	CLEAN	0																		
L	ZONAL	ZONAL RATING																		
	LOCATIO	LOCATION FACTOR					j													
1	WEIGHTE	WEIGHTED RATING	43	43.75	100	100.00	ř	36.25		7.50	100.00	00.	88	88.75	2	2.50	13.50	50		7.50

STANDARD COMPUTATION SHEET FOR PISTON RATING

PISTON NO. _

RATER IYONS DATE 9 July 1980
LABORATORY TEST NUMBER H.S.F. -1
STAND NO. 5 ENGINE NO. 6D-178671
FUEL AL-9242-F 1750F2

NO. 1 GROOVE, VOLUME-%
PISTON WTD* RATING
42

L															-	PISTO	PISTON WTD. RATING	RATIN	ပ	420.
						GRO	ROOVES							LA	LANDS				ONC	UNDER.
	DEPOSIT TYPE	FACTOR	NO.	1.1	ž	NO. 2	N	NO. 3	NO. 4	4	NO. 1	1	S	NO. 2	NO. 3	6	NO.	+	CROWN	N.
			AREA-%	AREA-% DEWERIT AREA-% DEN	AREA.%	DEMERIT	AREA.%	DEMERIT	AREA-%	DEMERIT	AREA-%	IERITAREA% DEMERIT AREA% DEMERITAREA% DEMERITAREA% DEMERITAREA% DEMERIT AREA.X DEMERITAREA.X DEMERITAREA.X DEMERIT	AREA-%	DEMERIT	AREA-X	DEMERIT	AREA.%	EMERIT	AREA.%	DEMER!
L	ЭН	1.00	1.5	15,00	100	100.00					100	100.00	06	00.06						
	MHC	0.75									Γ		1							
	S MC	0.50	85	42.50			20	10,00												
	27 27	0.25											9	2.50	6	25.00				
82	S VLC	0.15					80	12.00							Г					
 !	3€	CARBON	57	57.50	10	100.00	22	22.00			100	100.00	92.50	50	25	25.00				
	18 F	0.100							30	3.00										
	DBrL	0.075							70	5.25							8	7.50	100	7.50
92	K AL	0.060																		
	OU LAL	0.025																		
<u> </u>	VIAL.	0.010																		
<u></u>	A.	0.001																		
J	32	LACQUER							8.25	25							7.50	0	7.50	0.00
لــــا	CLEAN	0																		
	ZONAL	ZONAL RATING																		
لب	LOCATION	LOCATION FACTOR																		
	WEIGHTE	WEIGHTED RATING	57.	57.50	100.00	•00	22	22.00	8.25	25	100.00	8	92.	92.50	25.00	8	7.50	0	7.50	g
l												-						,		2

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AREAMDEMERITAREAMDEMERITAREAMDEMERITAREAMDEMERITAREAMDEMERIT LANDS 75 25 2.50 90 90.00 STANDARD COMPUTATION SHEET FOR PISTON RATING NO. 2 92.50 _DATE 9 July 1980 10 100 100.00 H.S.F. -1 100.00 ENGINE NO. 6D-178671 N.0 AREA% DEMERIT AREA% DEMERIT AREA% DEMERIT AREA% DEMERIT 7.50 LABORATORY TEST NUMBER_ NO. 4 100 LYONS 18.75 FUEL AL-9242-F 12.50 31.25 STAND NO. 5 NO. 3 RATER___ GROOVES 75 25 100 100.00 100.00 NO. 2 17.50 15.00 32.50 N 0. TEST LABORATORY AFIRE LUBRICANT AL-9065-L TEST PROCEDURE 240 30 20 DEPOSIT FACTOR 0.18 0.075 0.060 FEST HOURS_ 8. 0.75 0.15 0.50 0.25 CARBON DBrL LACQUER P DEPOSIT TYPE MHC VLC Ş Ž 8 Ç ပ္ A

UNDER-CROWN

NO.3

403

NO. 1 GROOVE, VOLUME-% PISTON WTD. RATING

R-2

PISTON NO.

83

CARBON

0.025 0.010 0.00

7.50

100

3.00

å 3

22.50

18.75 3.75 7.50

9.00

7.50

7.50

9.00

22,50

92,50

100,00

7.50

31,25

100.00

WEIGHTED TOTAL DEPOSITS

LOCATION FACTOR WEIGHTED RATING

ZONAL RATING

0

CLEAN

LACQUER

R

Bearing of Torrest of the second second of the second seco

STANDARD COMPUTATION SHEET FOR PISTON I'ATING
RATER LYONS DATE (9 July 1980 PISTON NO. R-3
LABORATORY TEST NUMBER H.S. F.-1
STAND NO. 5 ENGINE NO. 60-178671

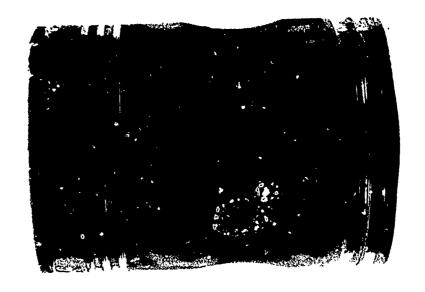
12SDF-2

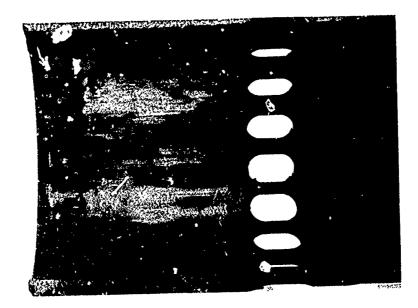
FUEL_AL-9242-F

TEST LABORATURY AFIRE LUBRICANT AL-9665-L

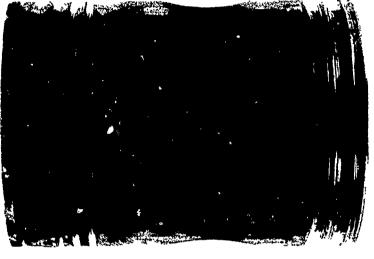
NO. 1 GROOVE, VOLUME-%

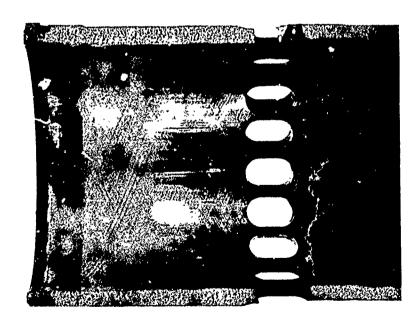
							Ì								1	SMILE DIM NOISE	1		,	CCC
L						GRO	ROOVES							LA	LANDS	,			3	UNDER.
	DEPOSIT	DEPOSIT FACTOR	NO.	-	ž	NO. 2	Ž	NO. 3	ž	NO. 4	NO.	-	NO. 2	2	NO.3	ند	Š.	+	5	CROWN
	4		AREA-% DEMERIT AREA-% DEM	MERIT	AREA-X	DEMERIT	AREA-%	DEMERIT	AREA-9	IERITAREA% DEWERITAREA% DEWERITAREA% DEWERITAREA% DEWERITAREA% DEWERIT AREA % DEWERIT AREA % DEWERIT	AREA-%	CEMERIT	AREA-%	EMERIT	AREA-X	LEMERIT	AREA.%	DEMERIT	AREA-%	DEMERIT
<u> </u>	웃	1.00			100	100.00					100	100.00	85	85.09						
	N:HC	0.75																		
NO	₹	0.50																		
8A/	၁၂	0.25	100	25.00			20	5.00					15	3.75						
/ <u>)</u> 8/	V.C	0.15					80	12.00							65	9.75				
	J ∉	CARBON RATING	25.00	90	100.00	.00	1	17.03			100.00	00	88.75	75	9.75	75				
	18	G. 100													15	1.50	40	4.00		
	DBrL	0.075													20	1.50			100	7.50
H :	AL:	0.050							99	5.00							909	3.00		
ino	O) LAL	0.025																		
DA.	3	0.010																		
	2	0.001																		
1	2=	LACQUER RATING							2	5.00					3.00	00	7.00	00	7.	7.50
	CLEAN	0																		
	ZONAL	ZONAL RATING																		
	.0CA110	LOCATION FACTOR																		
_	WEIGHTE	WEIGHTED RATING	25.00	0	10	100.00	-	17.00	5	5.00	100	100.00	88	88.75	3.00	2	7.00	8	1	7.50
J																1				





NO. 3-R THRUST SIDE (BEST)

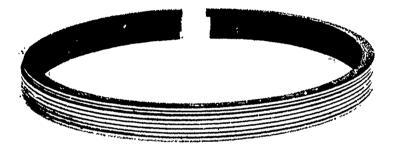




RING FACE CONDITION TEST: HSF-1



Piston 1-R



Piston 2-R



Piston 3-R

RING FACE CONDITION TEST: HSF-1



Piston 1-L



Piston 2-L



Piston 3-L

APPENDIX E

240-HOUR TRACKED-VEHICLE CYCLE USING 6V-53T DIESEL ENGINE

Test Lubricant: AL-8980-L Test Fuel: 1% S DF-2 Engine Test Number: HSF-2 Date Completed: 6 August 1980

Conducted For

U.S. Army Mobility Equipment Research and Development Command
Energy and Water Resources Laboratory
Ft. Belvoir, Virginia

bу

U.S. Army Fuels and Lubricants Research Laboratory Southwest Research Institute San Antonio, Texas 78284

6V-53T TEST: HS-2

ENGINE REBUILD MEASUREMENTS Model Number: 5063-5397 Serial Number: 6D-178671

	Min	<u>Max</u>	Avg	Specified Limits
Cylinder Block Bore Inside Diameter (bottom) Out-of-Round Taper	4.3572 0.0000 0.0000	4.3578 0.0006 0.0002	4.3575 0.0003 0.0001	4.3595 max 0.0015 max 0.0015 max
Cylinder Liners (installed) Inside Diameter Out-of-Round Taper	3.8755 0.0000 0.0000	3.8765 0.0008 0.0009	3.8761 0.0003 0.0003	3.8752-3.8767 0.0020 max 0.0010 max
Piston Diameter (@ skirt)	3.8679	3.8684	3.8681	3.8669-3.3691
Piston Skirt to Cylinder Liner Clearance	0.0074	0.0085	0.0080	0.0061-0.0098
Compression Rings Gap (Fire) Gap (Others)	0.030 0.032	0.043 0.045	0.036 0.039	0.0200-0.0460 0.0200-0.0460
Ring-to-Groove Clearance Top (Fire) No. 1 No. 2 and 3	0.004 0.007 0.005	0.004 0.009 0.007	0.004 0.008 0.006	0.003-0.006 0.007-0.010 0.005-0.008
Oil Control Rings Gap Ring-to-Groove Clearance	0.019 0.0015	0.022 0.005	0.020 0.003	0.010-0.025 0.0015-0.0055
Piston Pin Pin-to-Piston Bushing Clearance Pin-to-Conn. Rod Bushing Clearance	0.0029	0.0034 0.0017	0.0032	0.0025-0.0034 0.0010-0.0019
Connecting Rod Bearing- to-Journal Clearance	0.0021	0.0028	0.0028	0.0011-0.0041
Main Bearing-to-Journal Clearance	Bearings 1	replaced but	not measure	ed 0.0010-0.0040
Camshaft Bearing-to- shaft Clearance	Bearings n	not replaced	or measured	i 0.0045-0.0060

6V-53T

240-HOUR TRACKED VEHICLE CYCLE ENDURANCE TEST

TEST: HSF-2

OPERATING CONDITIONS SUMMARY

Lubricant: AL-8980-L Fuel: 1% S DF-2 (AL-9697-F)

	Maximum Po (2800 r	pm)	Maximum Tor	rpm)
	<u>Mean</u>	Standard * Deviation	Mean	Standard Deviation
Engine Speed, rpm	2805	5	2202	7
Torque, Ft-1b (N-m)	572 (776)	14(19)	648(879)	14(19)
Fuel Consumption, 1b/hr (kg/hr)	121.5(55.16)	3.3(1.50)	107.3(48.71)	1.7(.77)
Observed Power, Bhp (kW)	305 (228)	8(6)	272(203)	6(4)
BSFC, 1b/Bhp-hr (g/kW-hr)	0.399(243)	0.006(4)	0.395(240)	0.005(4)
Temperatures, °F (°C)				
Exhaust before turbo	989 (532)	30(17)	1032 (556)	33(18)
Exhaust after turbo	813(434)	15(8)	851 (455)	16(9)
Water Jacket Inlet	156 (69)	1(1)	156 (69)	1(1)
Water Jacket Outlet	170 (77)	1(1)	170(77)	1(1)
Oil Sump	237 (114)	5(3)	228(109)	5(3)
Fuel at Filter	92(33)	2(1)	90(32)	1(1)
Inlet Air (at compressor)	99 (37)	5(3)	98 (37)	5(3)
Airbox	293(145)	5(3)	243(117)	5(3)
Pressures				
Exhaust before turbo, psi (kPa)	14.6(101)	0.6(4)	10.1(70)	0.6(4)
Exhaust after turbo, in. Hg (kPa	a) 2.7(9.2)	0.1(0.3)	1.8(6.1)	0.1(0.3)
Compressor Discharge, psi (kPa)	14.0(96)	0.6(4)	10.7(74)	0.4(3)
Blower Discharge, psi (kPa)	19.5(134)	0.6(4)	12.2(84)	0.4(3)
Oil Gallery, psi (kPa)	38(262)	4(28)	32(220)	5 (34)
Intake Vacuum, in. H ₂ O (kPa)	7.1(1.77)	0.2(0.05)	4.4(1.10)	0.1(0.02)
Ambient Conditions (both modes of operation				
Dry Bulb Temperature, °F (°C)	79(26)	2(1)		
Wet Bulb Temperature, °F (°C)	82(28)	3(2)		
Barometric Pressure, in. Hg (kPa	· •	0.11(0.37)		

^{*68%} of the values for a given variable occur within ± 1 standard deviation of the mean; 95% occur within ± 2 standard deviations.

Unscheduled Shutdowns:

^{16.5} Hrs: Fuel filters replaced due to clogging and the resulting loss of power.

^{25.5} Hrs: Oil sight glass broken; 18.7 lb. oil replaced.

⁴⁰ Hrs: Shaft for blower drive had splines stripped; replaced; oil changed.

⁸⁰ Hrs: Fuel filters replaced due to clogging and the resulting loss of power.

¹⁴⁰ Hrs: Fuel filters replaced due to clogging and the resulting loss of power.

^{157.5} Hrs: Leaking injection line repaired; oil changed due to fuel dilution.

^{195.5} Hrs: Fuel filters replaced due to clogging and the resulting loss of power.

6V-53T TEST: HSF-2 Lubricant: AL-8980-L

LUBRICANT ANALYSIS

	ASTM			Test Tin	e, Hours	
	Test <u>Method</u>	New <u>011</u>	_60_	120	180	240
Kinematic Viscosity						
@ 40°C, cSt	D 445	109.1	52.9	33.8	111.8	138.6
Kinematic Viscosity						-5510
@ 100°C, cSt	D 445	11.65	7.6	5.8	11.9	13.6
Viscosity Index	D 2270	93	107	114	95	93
Total Acid Number, mg KOH	/g D 664	2.3	1.8	1.9	2.5	3.0
Total Base Number, mg KOH	/g D 664	13.3	6.3	3.8	6.5	5.5
Pentane B Insolubles, wt%	D 893	0.03	ND*	0.46	ND	(1.19
Toluene B Insolubles, wt%	D 893	0.01	ND	0.37	ND	0.15
Flash Point, °C (°F)	D 92	223	168(334)	164(328)	206 (402)	218(424)
Density, gm/ml @ 16℃				,		
(60 °F)	D 287	0.903	ND	0.901	ND	0.911
Carbon Residue, wt%	D 524	2.10	ND.	1.97	ND	2.44
Sulfated Ash, wt%	D 874	1.78	מיי	1.39	ND	2.01
Other		a				

^{*}ND = Not determined.

a = New oil contained: 0.48%w Ca

^{0.07%}w Zn 0.65%w S 0.07%w P

6V-53T TEST: HSF-2 Lubricant: AL-8980-L

TOTAL OIL CONSUMPTION AND WEAR METALS

		Wear Metal	s ⁺ , opm by XRF**
Test Time, Hours	Total Oil Consumed, 1b(kg)	Fe	Cu
10	-4.1(1.9)	_++	-
20	4.1(1.9)	33	ND*
25.5	18.7 lb. lost due to leak, n	ot counted to	wards
	oil consumed	•	
40	10.4(4.7)	44	16
40	Engine oil changed due to me	tal chips in	the oil
	caused by stripped blower dr		
60	12.3(5.6)	16	22
70	15.1(6.8)	-	-
80	17.1(7.8)	32	29
100	21.1(9.6)	59	32
120	26.2(11.9)	85	32
140	26.2(11.9)	28	10
156.5	Engine oil changed due to ex	cessive fuel	dilution
	from leaking injector; all p	rior oil cons	umption
	and wear metals figures inva		
160	28.6(13.0)	13	ND
173	33.4(15.1)	-	-
180	33.4(15.1)	18	ND
183	39.0(17.7)	6%a	-
200	43.8(19.9)	37	ND
210.5	50.6(23.0)	-	-
220	57.6(26.1)	43	28
230.5	63.1(28.6)	_	-
240	70.1(31.8)	46	10

Average Oil Consumption Rate, 0-240 hrs: 0.29 lb/hr (0.13 kg/hr) ** 160-240 hrs: 0.52 lb/hr (0.24 kg/hr)

⁺No other wear metals detected.

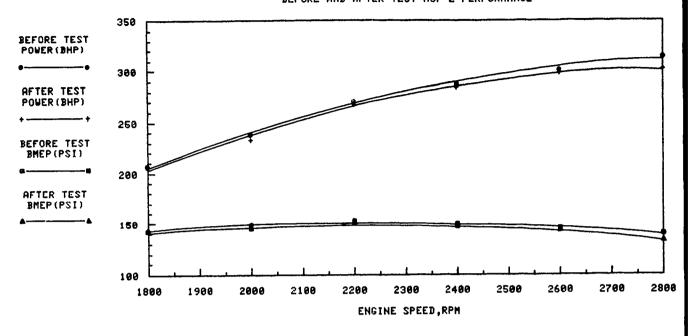
⁺Oil samples for wear metal analysis not taken at these times.

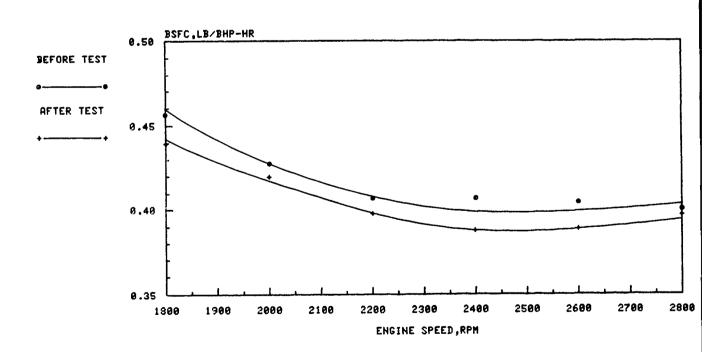
^{*}Not detected

^{**}Considerable error was introduced into this test data by the oil lost at 25.5 hrs and by the fuel dilution and subsequent oil change at 156.5 hrs.

6Y-53T 240 HOUR TRACKED VEHICLE CYCLE

BEFORE AND AFTER TEST HSF-2 PERFORMANCE





6V-53T TEST: HSF-2 Lubricant: AL-8980-L

WEAR MEASUREMENTS

Cylinder Liner Bore Diameter Change*

			Cylinder	Number		
	<u>1L</u>	_	_2	<u>L</u>	_3L	<u></u>
	$\underline{T-AT}^{**}$	<u>F-B</u>	T-AT	<u>F-B</u>	T-AT	<u>F-B</u>
Top	+0.0009	+0.0001	+0.0014	+0.0006	+0.0020	0.0000
Middle	+0.0008	+0.0004	+0.0007	+0.0009	+0.0005	+0.0003
Bottom	+0.0005	+0.0002	+0.0003	+0.0005	+0.0003	+0.0009
			Cylinder	Number		
	<u>1R</u>	•	_2	<u>R</u> _	_3R	_
	T-AT	<u>F-B</u>	T-AT	<u>zB</u>	<u>T-AT</u>	<u>F-B</u>
Тор	+0.0020	+0.0004	+0.0014	-0.0002	+0.0020	+0.0010
Middle	+0.0009	+0.0002	+0.0007	+0.0004	+0.0006	+0.0007
Bottom	+0.0006	+0.0002	+0.0003	+0.0007	+0.0004	+0.0001
			Average	Change		
			T-AT	F-B		
	Top		+0.0016	+0.0003		
	Midd	le	+0.0007	+0.0005		
	Bott	om	+0.0004	+0.0004		

Overall Average Change: +0.0007 in.

Piston Ring End Gap Change

Ring Number	<u>1L</u>	2L	_3L_	<u>1R</u>	<u>2R</u>	<u>3R</u>	Average Change
1	+0.002	+0.006	+0.003	+0.005	+0.004	+0.007	+0.005
2	+0.003	+0.004	+0.004	+0.003	-0.003	+0.005	+0.003
3	+0.001	+0.007	+0.003	+0.002	-0.004	+0.006	+0.003
4	+0,001	+0.006	+0.004	+0.002	0.000	+0.001	+0.002
5	+0.007	+0.004	+0.004	+0.005	+0.007	+0.008	+0.006
υ	+0.007	+0.003	+0.006	+0.006	+0.004	+0.006	+0.005
7	+0.003	+0.003	+0.007	+0.005	+0.003	+0.005	+0.004

Overall Average Change: +0.004 in.

*All dimensions given are in inches.

**T-AT = Thrust-Antithrust Direction; F-B = Front-Back Direction.

6V-53T TEST: HSF-2 Lubricant: AL-8980-L

POST TEST ENGINE CONDITION AND DEPOSITS

A. Cy	linde	r Liner
-------	-------	---------

A. Cylinder Liner			Cylind	er Number			
	1L	<u> 2L</u>	3L	<u> 1R</u>	<u>2R</u>	<u>3R</u>	Average
Intake Port Plugging, % restriction	5	3	2	2	3	2	3
Liner Scuffing, % Area Thrust Anti-Thrust Total	5 5 5	4C 10 25	20 5 13	10 10 10	5 5 5	5 15 10 Overall:	14 8 11
Liner Glazed, % Area	15	10	15	10	10	10	12
Liner Lacquer, % Area	85	90	85	90	90	90	88
B. Pistons Ring Face Burn, % Area Fire Ring No. 1 No. 2 No. 3	5 7 2 5	10 75 65 65	15 35 20 25	5 35 10 25	5 15 35 45	5 70 85 45 Overall:	8 40 36 35 30
Ring Groove Carbon, % Filling Fire Ring No. 1 No. 2 No. 3	5 70 15 1	5 75 20 2	20 85 15 0	10 90 20 0	5 85 10 0	5 75 2 0 Overall:	8 80 14 1
Piston Skirt Deposit Rating** (Demerit) Thrust Anti-Thrust	6.2 LS,LSC 6.0 LS	6.4 LS LSC 6.2 LS	6.0 PM,SC 6.0 LS	6.2 PM,SC 6.2 LS	6.4 PM,SC 6.2 LS	6.0 LS,LSC 6.2 LS Overall:	6.2 6.1 6.2

 $[\]star_{L}$ = Light, S = Scratches, PM = Plating Melted, N = Normal, SC = Scuffing. \star_{0} = least, 9 = most.

• -----,

6V-53T

TEST: HSF-2

Lubricant: AL-8980-L

POST TEST ENGINE CONDITION AND DEPOSITS

Other Ratings

			Cylinde	r Number		
	<u>1L</u>	<u>2L</u>	_3L_	<u>1R</u>	2R	<u>3R</u>
Tappets, Cams,						
Rocker Arms						
Tappet Deposit			Clean			
Tappet Surface						
Condition	All had	dull fin:	ish, unlike	normally	encount	ered high polish*
Cam Lobes	SCR**	SCR	N	SCR	SCR	SCR
Rocker Arms						
Tip			Normal			,
Bushing			Normal			2 cru-000 -darline 100-day crus
Shaft			Normal			, c.
Bearing Surface			,			
Condition	<u>No. 1</u>	No. 2	No. 3	No. 4		
Main Bearings and						
Main Journals	SCR	SCR	SCR,WI	SCR		
Rod Bearings	SCR	SCR	SCR	LSCR	N	LSCR
Rod Journals	LSCR	LSCR	LSCR	LSCR	LSCR	LSCR
Piston Pin	N	VLSC	SC,SCR	SC,SCR	SC	VLSC
Pin Bushing	MW***	MW	MW	HW	HW,G	LW
Combustion Chamber						
<u>Deposits</u>						
D			5%			
С			5%	10%		15%
В	10%	5%	10%	10%	50%	303
A	15%	1 0%	20%	15%	15%	20%
¹ 4A	75%	85%	60%	65%	35%	35%
Valve Covers, Cylinder						
Head Decks, Oil Pan			All C1	ean		

^{*}Possibly an etching of some sort.

**SCR = scratched, WI = wiping.

***L = light, M = medium, H = heavy, W = wear, G = gauling.

POST TEST ENGINE CONDITION AND DEPOSITS

Cylinder Number

B. Pistons (continued)

	<u>1L</u>	<u>2L</u>	_3L	<u>1R</u>	2R	<u>3R</u>	Average
Oil Control Ring Grooves (Demerit)							
Upper	5.0	5.0	5.0	5.0	5.0	5.0	5.0
Lower	5.0	5.0	5.0	5.0	5.0	5.0	5.0
						Overall:	5.0
Piston Groove Inside Diameter, % Ring Supporting Carbon						overali.	3.0
No. 1	0	0	0	0	0	0	0
No. 2	29	61	49	49	30	48	44
						Overall:	22
Piston WTD* Rating	390	420	395	319	323	342	`365
Ring Sticking No. 1 No. 2 No. 3 No. 4 Piston Oil Drain Holes C. Exhaust Valves Deposits Head Face			16	All Free All Free All Free OO% Open			
Tulip			% AHC t	o soot			
•		·	BHC to	% soot			
Stem			- No. 9 [™] Ia	cquer to	clean		
Surface Condition	_		_	_			
Freeness in Guide	F	F	F	F	F	F	
Head	N	N	N	N	N	N	
Face			-Light pit				
Seat	N	N	N	N	N	N	
Stem	N	N	N	N	N	N	
Tip	N	N	N	N.	N	N	

^{*}CRC Weighted Total Deposits (0 = least, 900 = most).

^{**}P = Pinched, F = Free, S = Stuck.

^{*}HC = Hard Carbon; the number-letter prefix indicates carbon depth with 12 = least, to J = most.

⁺⁺ = The higher the number, the darker the lacquer. (0 = lightest, 9 = darkest).

STANDARD COMPUTATION SHEET FOR PISTON RATING LABORATORY TEST NUMBER HSF-2 STAND NO STAND NO. 5 ENGINE NO. 6D-178671 FUEL AL-9697-F, 1% S DF-2 TEST LABORATORY AFIRE LUBRICANT AL-8980-L TEST PROCEDURE 240 TEST HOURS_

PISTON NO. 1-1

NO. 1 GROOVE, VOLUME-%
PISTON WTD* RATING 390

															SONA					EB.
L			-			GROO	OOVES							- 1				Ţ.	CHOMA	Ş
		Tioner				. 1	CN CN	~	4 CN	4	NO.	_	NO. 2	.2	80. 3		2	•		
<u> </u>	TYPE	FACTOR	2	-	2		2	,	THE WILLIAM OF A PER MINE WEEKIT AREA MOEMERIT AREA MOEMERIT	FMCBIT	ABEAKI	FWERIT	AREA-%	DEMERIT	AREA-XO	EMERIT	AREA-% DEMERIT AREA-% DEMERIT	EMERIT /	INEA%	EMERIT
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	9:	8			080	80.00					85	85.00	85	85.00	3	3	+			
	٢	3.			3															
	MHC	0.75													25	12.50				
NC	3	0.50			20	10.00								1	5	1.50				
) BF	_L_	0.25	13	25.00			25	6.25			15	3.75	51	3.72	2		+	00		
CAI	Ľ		3				75	11.25	50	7.50						1	3	3		
100	:	–l ₹				-	;	6	7 50	20	88	88.75	88	88.75	Ñ	50.00	9.	9.00		
		RATING	2	25.00	8	90.00	7	200	1	3										
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	¥ ₹	0.010			_		\downarrow		1											
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	-	MAIIMO	1		-										_					
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	10Z	ZONAL RATING					-						1		1					
	LOCAT	LOCATION FACTOR	_						- -	8	88	26 75	100	88.75	2(50.00	12	12.00	7	7.50
	WEIGH	WEIGHTED RATING		25.00	5	90.06		17.50		10.00			-							
_																				

STANDARD COMPUTATION SHEET FOR PISTON RATING RATER Ed Lyons DATE 8-7-80

LABORATORY TEST NUMBER HSF-2 STAND NO. 5 ENGINE NO. 6D-178671 FUEL AL-9697-F, 1% S DF-2 TEST LABORATORY AFIRE LUBRICANT AL-8980-L TEST HOURS 240 TEST PROCEDURE_

PISTON NO.

L-2

420 NO. 1 GROOVE, VOLUME-% PISTON WTD. RATING

																			1	
L						GROOVES	VES							LA	LANDS				3	UNDER.
ō	DEPOSIT	DEPOSIT	Ç	-	Ž	NO. 2	S	NO. 3	NO. 4	4	NO. 1	-	Ž	NO. 2	NO.3	3	Z	NO. 4	SR	Z Z
	TYPE	FACTOR	AREA-X	AREA-% DEMERIT AREA-% DEMI	AREA-%	DEMERIT	AREA-%	DEMERIT	AREA.%	DEMERIT	AREA-%	DEMERIT	AREA-%	ERITAREAS DEMERIT AREAS DEMERITAREAS DEMERITAREAS DEMERITAREAS DEMERITAREAS DEMERITAREAS DEMERITAREAS DEMERITAREAS	AREA-X	SEMERIT	AREA-%	DEMERIT	AREA-%	DEMERIT
	Ŭ H	180			75	75.00	20	20.00			96	90.06	85	85.00						
	Z HC	0.75																		
NC		0.50			25	12.50	20	10.00							70	35.00				
เลล	1	0.25	100	25.00			. 09	15.00			10	2.50	15	3.75	30	7.50	25	6.25		
	VLC	0.15							9	9.00							65	9.75		
01	3€	CARBON	25	25.00	8	89.50	45	45.50	9.	9.00	92	92.50	8	88.75	42	42.50	16	16.00		
	ם	0.100																		
	DBrL	<u> </u>							40	3.00							10	.75	100	7.50
8	4	0.060																		
31 IC	ZOE	0.025																		
<u> </u>	3	0.010																		
<u>!</u>	2	0.001																		
	2.	LACQUER							ė [3.00								.75	7	7.50
ــــــــــــــــــــــــــــــــــــــ	CLEAN	0																		
L	ZONAL	ZONAL RATING																		
ــــــــــــــــــــــــــــــــــــــ	LOCATIO	LOCATION FACTOR	-																	
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STANDARD COMPUTATION SHEET FOR PISTON RATING

TEST PROCEDURE 240

TEST HOURS 240

TEST LABORATORY TEST NUMBER HSF-2

TEST LABORATORY TEST NUMBER HSF-2

STAND NO. 5 ENGINE NO. 6D-178671

FUER CANT AL-8980-L

FUER CANT BY STAND NO. 5 ENGINE NO. 6D-178671

PISTON NO. L-3

NO. 1 GROOVE, VOLUME-%
PISTON WTD* RATING 394

L															-	PISTO	N WTD	PISTON WTD. RATING	9	394
	9	-			1	GRO	ROOVES							LA	LANDS				S	UNDER.
_J	DEPUSIT TYPE	FACTOR	NO.	1.1	ž	NO. 2	ž	NO. 3	ၟႄ	NO. 4	NO. 1	.1	N	NO. 2	Ž	NO. 3	ž	¥.0×	2	CROWN
			AREA-%	AREA-% DEMERIT AREA-% DEM	AREA.%		AREA-%	DEMERIT	AREA-%	DEMERIT	AREA-%	DEMERIT	AREA-%	DEMERIT	AREA.%	DEMERIT	AREA-%	ERITAREAS DEMERIT AREAS DEMERIT AREAS DEMERITAREAS DEMERITAREAS DEMERITAREAS DEMERITAREAS DEMERITAREAS DEMERIT	AREA-%	DEMER
	오	1.00			100	100.00					95	95.00	7.5	75.00						,
	MHC	0.75																		
NU	WC ON	0.50	55	27.50			35	17.50												
7 50	ပ	0.25	45	11.25			9	16.25			5	1.25	25	6.25	20	12.50	5	1.25		
10	VLC	0.15													50	7.50	15	2.25		
	۵ ش	CARBON Rating	38	38.75	100	100.00	3.	33.75			96	96.25	8	81.25	20	20.00	3.	3.50		
	81	0.100																		
	DBrL	0.075							100	7.50							80	9.00	100	7.50
83	AL.	0.050																		
סחו	LAL C	0.025																		
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	ZONAL	ZONAL RATING																		
لتا	OCATIO	LOCATION FACTOR																		
لــًــا	WEIGHTE	WEIGHTED RATING	38	38.75	100	100.00	33.	33.75	7.50	o O	96.25	25	81.25	25	20.00	8	9.50	0	7.50	20
į																				

A STANDER CONTINUED CONTINUED CONTINUED CONTINUED FOR CONTINUED CO

STANDARD COMPUTATION SHEET FOR PISTON RATING DATE 8-8-80 HSF-2 STAND NO. 5 ENGINE NO. 6D-178671 FUEL AL-9697-F: 12 S DF-2 LABORATORY TEST NUMBER_ RATER Ed Lyons

TEST LABORATORY_AFIRI__

TEST PROCEDURE 240

TEST HOURS_

PISTON NO.

3980-1. FUEL AL-9 3980-1. GROOVES NO. 1 NO. 2 NC AREA-% DEMERITAREA-% 100 100.00 20	AL-9697-F: 12 ES	7. S. DF-2				PISTON	PISTON WTD* RATING	N.	
GROOVES NO. 2 NC A-% DEMERITAREA-% DO 100.00						PISTON	WTD. RAT	SM1	
GROOVES NO. 2 NC A-X-DEMERITAREA-X DO 100.00								2	319
NO. 2 NG A-X DEMERITAREA-X DO 100.00	-				LANDS			5	UNDER.
A-% DEMERITAREA-% DO 100.00	۳.0 ص	NO. 4	NO. 1	NO. 2	NO.3	3	NO. 4	5	CROWN
00 100.00	DEMERIT AF	EA-XOEMERIT	REA-X DEMERIT AREA-X DEMERIT AREA-X DEMERIT AREA-X DEMERIT AREA-X DEMERIT AREA-X DEMERIT AREA-X DEMERIT	AREA-% DEMER	IT AREA'SO	EMERIT	AREA-S DEMER	IIT AREA!	DEMERIT
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WEIGHTED TOTAL DEPOSITS

STANDARD COMPUTATION SHEET FOR PISTON RATING _DATE 8-8-80 HSF-2

STAND NO. 5 ENGINE NO. 6D-178671 FUEL AL-9697-F: 17S DF-2 LABORATORY TEST NUMBER_ RATER Ed Lyons TEST LABORATORY AFIRIL LUBRICANT AL-8980-L TEST PROCEDURE 240 TEST HOURS__

PISTON NO.

NO. 1 GROOVE, VOLUME-% PISTON WTD. RATING

1															-					273
		-				GROC	ROOVES	;						LA	LANDS				S	ER.
	DEPOSIT	DEPOSIT	Ž	NO. 1	Ž	NO. 2	ž	NO. 3	NO. 4	. 4	NO.	. 1	NO. 2	.2	NO. 3	.3	Š.	4	S	CROWN
		5	AREA-%	AREA-% DEMERIT AREA-% DEM	AREA.N		AREA-%	ERITAREA% DEMERIT AREA% DEMERIT	AREA-%	DEMERIT	AREAK	DEMERIT	AREA-K	AREAM DEMERITAREAM DEMERITAREAM DEMERITAREAM DEMERITAREAM DEMERIT	AREA-X	SEMERIT	AREA-X	EMERIT	AREA-%	DEMERIT
K	웃	1.8			85	85.00					56	95.00								
	₹	3.0.75			15	11.25														
	Ş NO	0.50					15	7.50					50	25.00	15	7.50				
	887 디	0.25					15	3.75			5	1.25		12.50	85	21.25				
10	S VLC	0.15	100	15.00			70	10.50									85	12.75		
)4		CARBON	15	15.00	6	96.25	21	21.75			96.25	25	37.50	50	28.75	75	12	12.75		
	ᆱ	0.100																٠		
	DB 1	- 0.075							70	5.25									100	7.50
	R AL	0.050							30	1.50								`		
	S LAL	0.025															15	.375		,
	3 ₹	0.010																		
*******	ا ا	0.001																	·	
		LACQUER							6.75	75						٠	.3	375	7.	7.50
	CLEAN	0																. **	,	
	ZONA	ZONAL RATING											`							
	LOCATIC	LOCATION FACTOR										1					Ì			4.7.
	WEIGHT	WEIGHTED RATING	15	15.00	6	96.25	21.	21.75	6.75	75	96.25	25	37.50	50.	28.75	75	13.125	125	7.50	0
•	1013111	TOT GUTIO	100	32130030																

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STANDARD COMPUTATION SHEET FOR PISTON RATING

PISTON NO. -

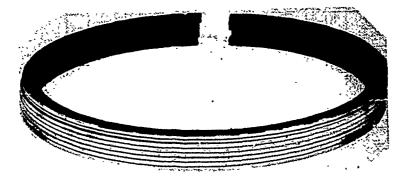
TEST PROCEDURE 240
TEST HOURS 240
TEST LABORATORY AFLRL
LUBRICANT AL-8980-L

RATER Ed Lyons
LABORATORY TEST NUMBER HSF-2
STAND NO. 5 ENGINE NO. 6D-178671
FUEL AL-9697-F; 1% S DF-2

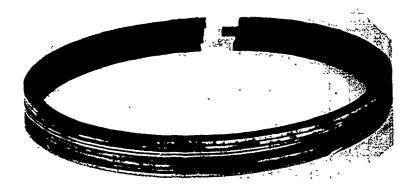
NO. 1 GROOVE, VOLUME-%
PISTON WTD* RATING
342

												 :	9					
				Ē	ROOVES	S						5	LANDS				CNDEK	F.
DEPOSIT		CN		NO. 2	\vdash	NO. 3	_	NO. 4	N	NO. 1	×	NO. 2	Š	6	Š	*	SHO	2
TYPE	FACTOR	AREAS DEWERTTAREAS DEMERITAREAS DE DEMERITAREAS DEMERITAREAS DEMERITAREAS DEMERITAREAS DE DE DEMERITAREAS DE	ERIT ARE	-XDEM	ERITAR	EA-% DEME	RIT ARE	N-% DEMER	TAREA.	DEMERIT	AREA.%	DEMERIT	AREA-%	EMERIT	AREA-%	EMERIT	AREA.X	DEMERIT
77	9		50	50.0	8	-	_	_	75	75.00	55	55.00	30	30.00				
N N	٠		20	1	50													
NC NC			-		-	5 2.50	0								2	5.00		
, 88 고	+					5 1.25	5.		25	6.25	45	11.25	2	5.00	15	3.75		
VC VC	C 0.15	100 15	15.00		6	90 13.50	0	_					20	7.50	65	9.75		
<u> </u>	CARBON	15.00		87.50		16.75			8	81.25	99	66.25	42	42.50	18	18.50		
9	0.100		-	-	-		15	5 1.50										
08r	-														2	:75	100	7.50
A A	┿		_		-		85	5 4.25										
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<u> </u>	LACQUER							5.75		-					27.			7.50
CLEAN	0			\dashv			\dashv		_	4								
2	ZONAL RATING						\dashv		_		_							İ
7007	LOCATION FACTOR												_		!	30	7 50	c
XE.G	WEIGHTED RATING	15.00	_	87.50	-	16.75		5.75		81.25	9	66.25	42	42.50		19.25		

RING FACE CONDITION TEST: HSF-2



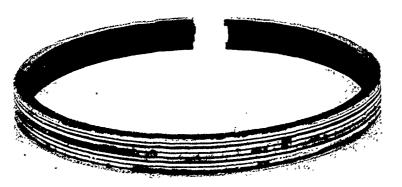
Piston 1-L



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Piston 2-L

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Piston 3-L

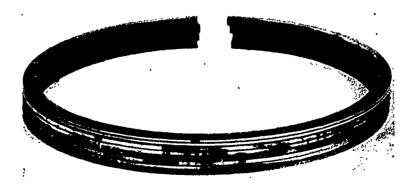


Piston 1-R



Piston 2-R

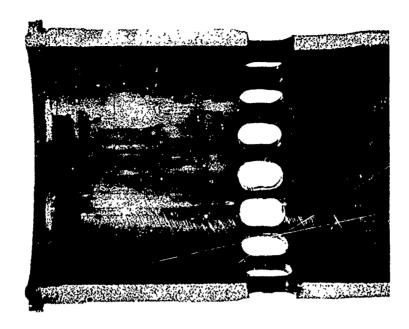
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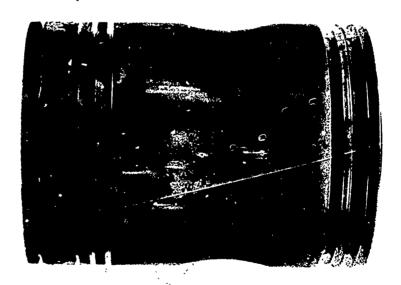


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Piston 3-R

PISTON AND CYLINDER LINER CONDITION TEST: HSF-2



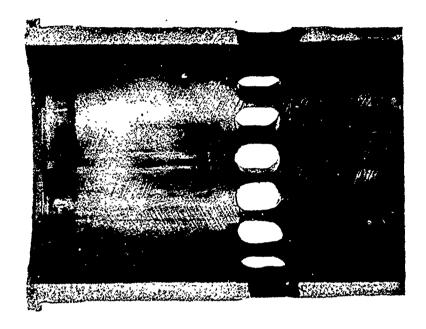


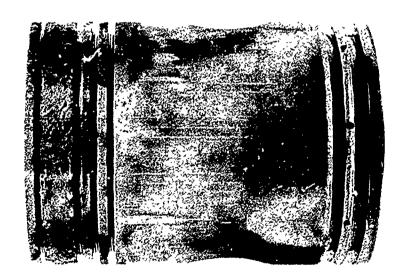
NO. 2-L THRUST SIDE (WORST)

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in the contraction of the contra

PISTON AND CYLINDER LINER CONDITION TEST: HSF-2





DISTRIBUTION LIST

DEPARTMENT OF DEFENSE		CDR US ARMY TANK-AUTOMOTIVE MATERIAL	
DEFENSE DOCUMENTATION CTR		READINESS CMD	
CAMERON STATION	12		1
ALEXANDRIA VA 22314			1
			1
DEPT OF DEFENSE	_	WARREN MI 48090	
ATTN: DASA (MRA&L) -ES (MR DYKEMAN)	1	D TO HAMAD	
WASHINGTON DC 20301		DIRECTOR	
COMMINER		US ARMY MATERIAL SYSTEMS ANALYSIS AGENCY	
COMMANDER			1
DEFENSE FUEL SUPPLY CTR ATTN: DFSC-T	1		1
CAMERON STA	1	DRXSY-L	1
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ALEAANDRIA VA 22314		indiana inotano dicottà in livos	
COMMANDER		CDR	
DEFENSE GENERAL SUPPLY CTR		US ARMY APPLIED TECH LAB	
ATTN: DGSC-SSA	1		1
RICHMOND VA 23297		DAVDL-ATL	1
		FORT EUSTIS VA 23604	
DEPARTMENT OF THE ARMY		•	
		HQ, 172D INFANTRY BRIGADE (ALASKA)
HQ, DEPT OF ARMY			I
ATTN: DALO-TSE	1	AFZT-DI-M DIRECTORATE OF INDUSTRIAL OPERATIONS	1
DAMA-CSS-P (DR BRYANT)	1	DIRECTORATE OF INDUSTRIAL	
DAMA-ARZ (DR CHURCH)			
DAMA-SMZ **	1	FT RICHARDSON AK 99505	
WASHINGTON DC 20310		CDD	
ODB		CDR US ARMY GENERAL MATERIAL &	
CDR U.S. ARMY MOBILITY EQUIPMENT		PETROLEUM ACTIVITY	
R&D COMMAND		•	1
Attn: DRDME-GL	10	STSGP-PE	î
FORT BELVOIR VA 22060	10		1
TONI DEDVOIR VII LEVOO		NEW CUMBERLAND ARMY DEPOT	_
CDR		NEW CUMBERLAND PA 17070	
US ARMY MATERIAL DEVEL&READINESS			
COMMAND		CDR	
ATTN: DRCLDC (MR BENDER)	1	US ARMY ARRCOM, LOG ENGR DIR	
DRCMM-SP (LTC O'CONNER)	1		1
DRCQA-E (MR SMART)	1	ROCK ISLAND ARSENAL IL 61299	
DRCDE-DG	1		
DRCRE-TF	2	CDR	
5001 EISENHOWER AVE		US ARMY COLD REGION TEST CENTER	
ALEXANDRIA VA 22333			1
400		APO SEATTLE 98733	
CDR		CND	
US ARMY TANK-AUTOMOTIVE MATERIAL		CDR	
READINESS CMD	1	US ARMY RES & STDZN GROUP (EUROPE)	
ATTN DRDTA-RG (MR HAMPARIAN) DRDTA-NS (DR PETRICK)	1	•	1
DRDTA-J	1	BOX 65	•
WARREN MI 48090	•	FPO NEW YORK 09510	
HITTINGS IN TOOLS			

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ud, no www varvitou ven cun		CDK	
ATTN DRDAV-D (MR CRAWFORD)	1	US ARMY EUROPE & SEVENTH ARMY	
DRDAV-N (MR BORGMAN)	1	ATTN AEAGC-FMD	1
DRDAV-E (MR LONG)	ī	APO NY 09403	_
P O BOX 209	•	77403	
		DBA I MAR INAMEDIAN BRA I ANA	
ST LOUIS MO 63166		PROJ MGR, PATRIOT PROJ OFC	
		ATTN DRCPM-MD-T-G	Ţ
CDR		US ARMY DARCOM	
US ARMY FORCES COMMAND		REDSTONE ARSENAL AL 35809	
ATTN AFLG-REG (MR HAMMERSTROM)	1		
AFLG-POP (MR COOK)	1	CDR	
FORT MCPHERSON GA 30330		THEATER ARMY MATERIAL MGMT	
		CENTER (200TH)	
CDR		DIRECTORATE FOR PETROL MOMT	
US ARMY ABERDEEN PROVING GROUND		ATTN AEAGD-MM-PT-Q (MR PINZOLA)	Ţ
ATTN STEAP-MT	1	ZWEIBRUCKEN	
STEAP-MT-U (MR DEAVER)	1	APO NY 09052	
ABERDEEN PROVING GROUND MD 21005			
		CDR	
CDR		US ARMY RESEARCH OFC	
US ARMY YUMA PROVING GROUND		ATTN DRXRO-EG	1
	1	DRXRO-CB (DR GHIRARDELLI)	i
ATTN STEYP-MT	1		T
YUMA AR 85364		P O BOX 12211	
		RSCH TRIANGLE PARK NC 27709	
MICHIGAN ARMY MISSILE PLANT			
OFC OF PROJ MGR, XM-1 TANK SYS		DIR	
ATTN DRCPM-GCM-S	1	US ARMY R&T LAB	
WARREN MI 48090		ADVANCED SYSTEMS RSCH OFC	
112 10000		ATTN MR D WILSTED	1
MICHIGAN ARMY MISSILE PLANT		AMES RSCH CTR	-
PROG MGR, FIGHTING VEHICLE SYS		MOFFITT FIELD CA 94035	
ATTN DRCPM-FVS-SE	1		
WARREN MI 48090		CDR	
		TOBYHANNA ARMY DEPOT	
PROJ MGR, M60 TANK DEVELOPMENT		ATTN SDSTO-TP-S	1
ATTN DRCPM-M60-E	1	TOBYHANNA PA 18466	
WARREN MI 48090	_		
70070		DIR	
DDOG MCD M112/M112A1 DAMTIN		US ARMY MATERIALS & MECHANICS	
PROG MGR, M113/M113A1 FAMILY			
OF VEHICLES		RSC/I CTR	_
ATTN DRCPM-M113	1	ATTN L'XXMR	1
WARREN MI 48090		WATERTOWN MA 02172	
PROJ MGR, MOBILE ELECTRIC POWER		CDR	
ATTN DRCPM-MEP-TM	1	US ARMY DEPOT SYSTEMS CMD	
7500 BACKLICK ROAD	-	ATTN DRSDS	1
SPRINGFIELD VA 22150		CHAMBERSBURG PA 17201	-
OLITHGETEDD AN EXID		OIMEDERODURG EN 1/201	
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CFC OF PROJ MGR, IMPROVED TOW		CDR	
VEHICLE		US ARMY WATERVLIET ARSENAL	_
US ARMY TANK-AUTOMOTIVE R&D CMD		ATTN SARWY-RDD	1
ATTN DRCPM-ITV-T	1	WATERVLIET NY 12189	
WADDEN MT AROOD			

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US ARMY LEA ATTN DALO-LEP NEW CUMBERLAND ARMY DEPOT NEW CUMBERLAND PA 17070	1	DIRECTOR US ARMY RSCH & TECH LAB (AVRADCOM PROPULSION LABORATORY ATTN DAVDL-PL-D (MR ACURIO) 21000 BROOKPARK ROAD CLEVELAND OH 44135	M) 1
CDR US ARMY GENERAL MATERIAL & PETROLEUM ACTIVITY ATTN STSGP-PW (MR PRICE) SHARPE ARMY DEPOT LATHROP CA 95330	1	CDR US ARMY NATICK RES & DEV CMD ATTN DRDNA-YEP (DR KAPLAN) NATICK MA 01760	1
CDR US ARMY FOREIGN SCIENCE & TECH CENTER ATTN DRXST-MT1	1	CDR US ARMY TRANSPORTATION SCHOOL ATTN ATSP-CD-MS FORT EUSTIS VA 23604	1
FEDERAL BLDG CHARLOTTESVILLE VA 22901 CDR		CDR US ARMY QUARTERMASTER SCHOOL ATTN ATSM-CD-M	1
DARCOM MATERIAL READINESS SUPPORT ACTIVITY (MRSA) ATTN DRXMD-MS	1	ATSM-CTD-MS ATSM-TNG-PT (COL VOLPE) FORT LEE VA 23801	1
HQ, US ARMY T&E COMMAND ATTN DRSTE-TO-O	1	HQ, US ARMY ARMOR SCHOOL ATTN ATSB-TD FORT KNOX KY 40121	1
ABERDEEN PROVING GROUND, MD 2100 HQ, US ARMY ARMAMENT R&D CMD	5	CDR US ARMY LOGISTICS CTR ATTN ATCL-MS (MR A MARSHALL)	1
ATTN DRDAR-SCM-OO (MR MUFFLEY) DRDAR-TST-S DOVER NJ 07801	1	FORT LEE VA 23801 CDR US ARMY FIELD ARTILLERY SCHOOL	
HQ, US ARMY TROOP SUPPORT & AVIATION MATERIAL READINESS COMMAND		ATTN ATSF-CD FORT SILL OK 73503	. 1
ATTN DRSTS-MFG (2) DRCPO-PDE (LTC FOSTER) 4300 GOODFELLOW BLVD ST LOUIS MO 63120	1	CDR US ARMY ORDNANCE CTR & SCHOOL ATTN ATSL-CTD-MS ABERDEEN PROVING GROUND MP 21005	1
DEPARTMENT OF THE ARMY CONSTRUCTION ENG RSCH LAB ATTN CERL-EM P O BOX 4005 CHAMPAIGN IL 61820	1	CDR US ARMY ENGINEER SCHOOL ATTN ATSE-CDM FORT BELVOIR VA 22060	1
HQ US ARMY TRAINING & DOCTRINE CMD ATTN ATCD-SL (MAJ HARVEY) FORT MONROE VA 23651	1	CDR US ARMY INFANTRY SCHOOL ATTN ATSH-CD-MS-M FORT BENNING GA 31905	1

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CDR US ARMY AVIATION CTR & FT RUCKER ATTN ATZQ-D FORT RUCKER AL 36362	1	CDR NAVAL FACILITIES ENGR CTR ATTN CODE 1202B (MR R BURRIS) CODE 120B (MR BUSCHELMAN) 200 STONEWALL ST	1
DEPARTMENT OF THE NAVY		ALEXANDRIA VA 22322	
CDR NAVAL AIR PROPULSION CENTER ATTN PE-71 (MR MAGETTI) PE-72 (MR D'ORAZIO) P O BOX 7176 TRENTON NJ 06828	1	CHIEF OF NAVAL RESEARCH ATTN CODE 473 (DR R MILLER) ARLINGTON VA 22217 CDR NAVAL AIR ENGR CENTER	1
CDR NAVAL SHIP ENGINEERING CTR		ATTN CODE 92727 LAKEHURST NJ 08733	1
CODE 6101F (MR R LAYNE) WASHINGTON DC 20362 CDR	1	CDR NAVY FACILITIES ENGRG CMD CIVIL ENGR SUPPORT OFC CODE 15312A (ATTN EOC COOK)	1
DAVID TAYLOR NAVAL SHIP R&D CTR CODE 2830 (MR G BOSMAJIAN) CODE 2831	1 1	NAVAL CONSTRUCTION BATTALION CTR PORT HUENEME CA 93043	1
ANNAPOLIS MD 21402 JOINT OIL ANALYSIS PROGRAM - TECHNICAL SUPPORT CTR		CDR, NAVAL MATERIAL COMMAND ATTN MAT-08T3 (DR A ROBERTS) CP6, RM 606 WASHINGTON DC 20360	1
BLDG 780 NAVAL AIR STATION PENSACOLA FL 32508 DEPARTMENT OF THE NAVY HQ, US MARINE CORPS		CDR NAVY PETROLEUM OFC ATTN CODE 40 CAMERON STATION ALEXANDRIA VA 22314	1
ATTN LPP (MAJ SANBERG) LMM (MAJ GRIGGS) WASHINGTON DC 20380	1	CDR MARINE CORPS LOGISTICS SUPPORT BASE ATLANTIC	
CDR NAVAL AIR SYSTEMS CMD ATTN CODE 52032E (MR WEINBURG) CODE 53645	1 1	ATTN CODE P841 ALBANY GA 31704 DEPARTMENT OF THE AIR FORCE	1
WASHINGTON DC 20361 CDR NAVAL AIR DEVELOPMENT CTR		HQ, USAF ATTN RDPT (MR EAFFY) WASHINGTON DC 20330	1
ATTN CODE 60612 (MR L STALLINGS) WARMINSTER PA 18974	1	CDR US AIR FORCE WRIGHT AERONAUTICAL	
CDR NAVAL RESEARCH LABORATORY ATTN CODE 6170 (MR H RAVNER) CODE 6180 CODE 6110 (DR HARVEY) WASHINGTON DC 20375	1 1 1	LAB ATTN AFWAL/POSF (MR CHURCHILL) AFWAL/POSL (MR JONES) WRIGHT-PATTERSON AFB OH 45433	1
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USAF SAN ANTONIO AIR LOGISTICS CTR	
ATTN SAALC/SFQ (MR MAKRIS)	1
SAALC/MMPRR (MR ELLIOT)	1
KELLY AIR FORCE BASE, TX 78241	
CDR	
US AIR FORCE WRIGHT AERONAUTICAL	
LAB	
ATTN AFWAL/MLSE (MR MORRIS)	1
AFWAL/MLBT	1
WRIGHT-PATTERSON AFB OH 45433	
CDR	
USAF WARNER ROBINS AIR LOGISTIC	
CTR	
ATTN WR-ALC/MMIRAB-1 (MR GRAHAM)	1
ROBINS AFB GA 31098	
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